

# **THERMODYNAMIC THEORY**

# EQUATION OF STATE

## CHARLES' EQUATION

$$p_1 V_1 / T_1 = p_2 V_2 / T_2 = \text{constant}$$

## EQUATION OF STATE (IDEAL GAS)

$$pV = \text{constant } T$$

$$pv = \text{constant } T / \text{mass}$$

$$pv = RT$$

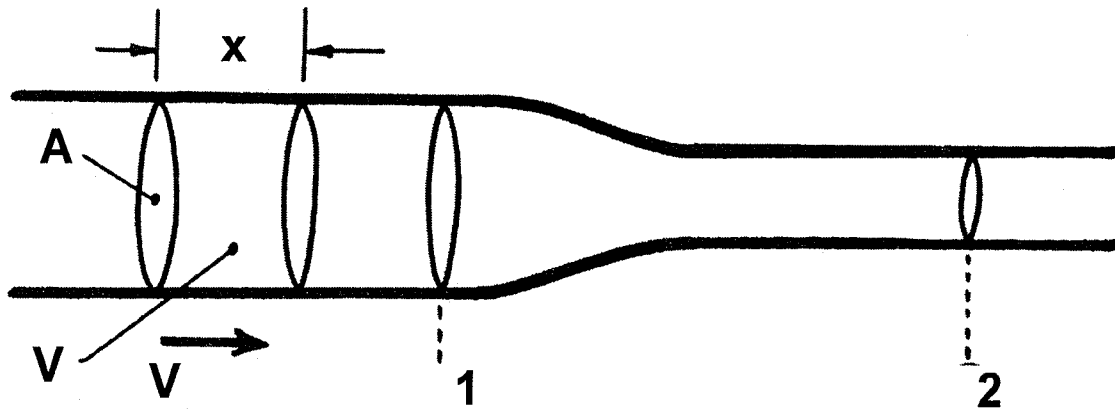
## EQUATION OF STATE (NON-IDEAL GAS)

$$pv = z R T$$

$$z = \text{compressibility factor}$$

# CONTINUITY EQUATION

(WHAT GOES IN..... MUST COME OUT)



Volume	$V = A x$	$(m^3)$
Density	$\rho$	$(kg/m^3)$
Mass	$m = \rho V = \rho A x$	$(kg)$
Velocity	$V = x / t$	$(m/s)$
	$x = V t$	
Mass	$m = \rho A V t$	$(kg)$
Mass Flow	$M = \rho A V$	$(kg/s)$

Continuity Equation states that:

$$\rho_1 V_1 A_1 = \rho_2 V_2 A_2$$

for steady mass flow rate of  $M$

# ENERGY EQUATION

THE FIRST LAW OF THERMODYNAMICS STATES THAT ENERGY IS NEITHER CREATED NOR DESTROYED...

Combining all types of energy into a single equation gives:

$$\begin{aligned} z_1 g + V_1^2 / 2 + u_1 + p_1 v_1 + w_{IN} + q_{IN} \\ = z_2 g + V_2^2 / 2 + u_2 + p_2 v_2 + w_{OUT} + q_{OUT} \end{aligned}$$

For constant  $u$  and no  $w$  or  $q$  this equation reduces to the Bernoulli Equation

$$z_1 g + V_1^2 / 2 + p_1 v_1 = z_2 g + V_2^2 / 2 + p_2 v_2$$

$$z_1 + V_1^2 / 2g + p_1 / \rho g = z_2 + V_2^2 / 2g + p_2 / \rho g$$

Note that:  $u + pv = h$

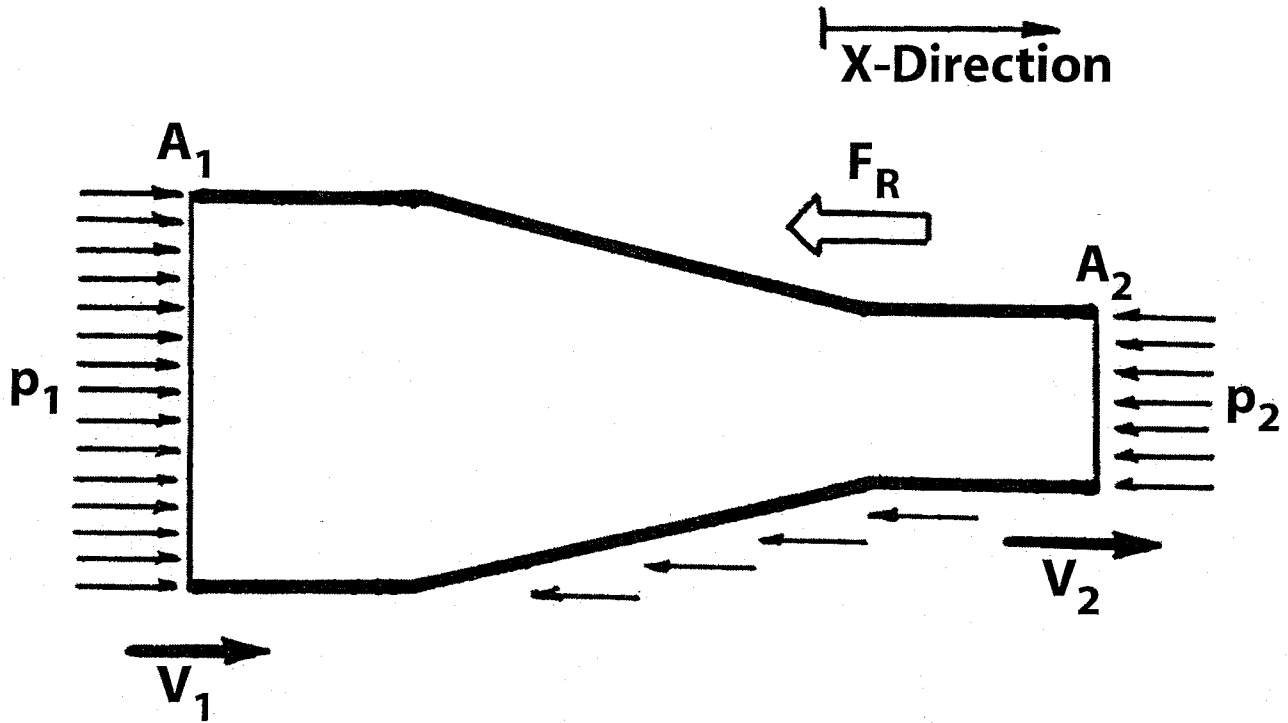
Hence rewriting gives:

$$\begin{aligned} (z_1 - z_2)g + (V_1^2 - V_2^2) / 2 + (h_1 - h_2) + (w_{IN} - w_{OUT}) \\ + (q_{IN} - q_{OUT}) = 0 \end{aligned}$$

This equation can be adapted to all flow processes by substituting appropriate values.

# APPLICATIONS

## CONICAL PIPE



**Forces in X - Direction (on fluid)**

$$\Sigma F_x = p_1 A_1 - p_2 A_2 - F_R = \rho Q \Delta V_x \dots$$

**Reaction**

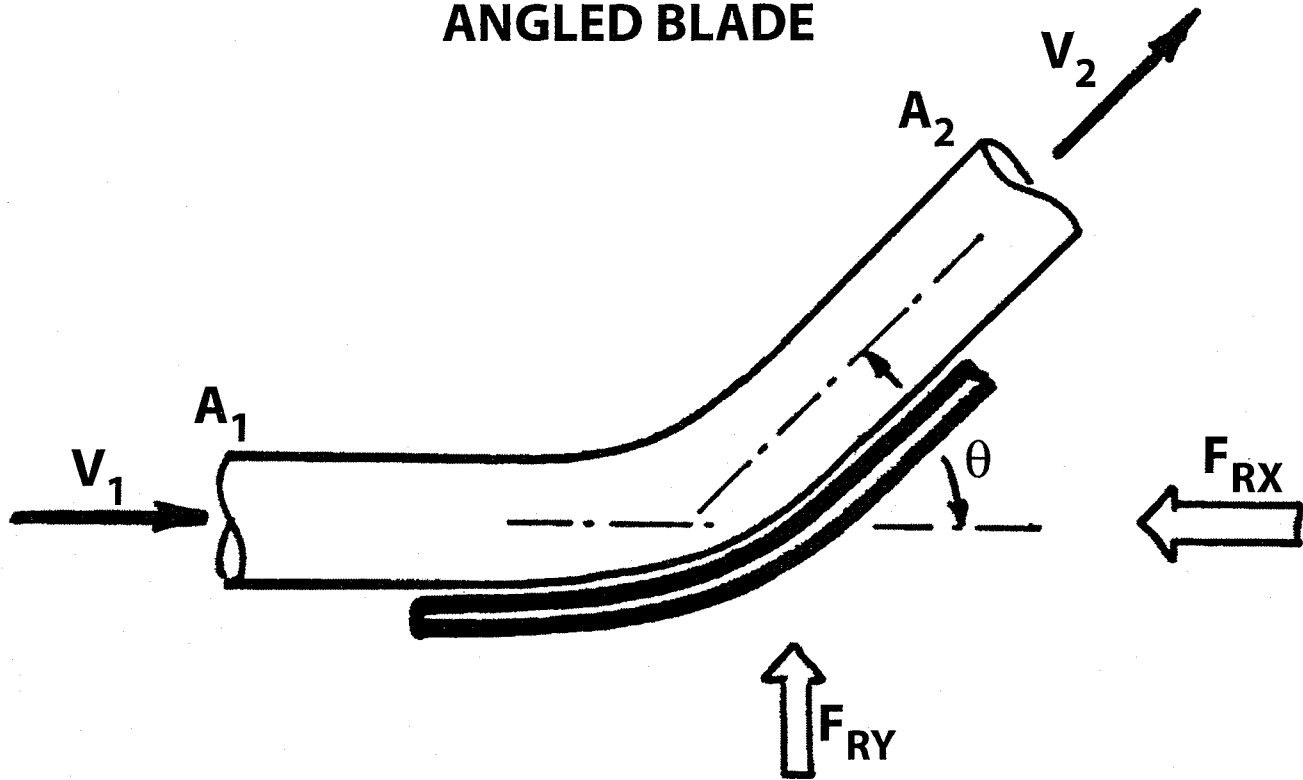
\* 
$$F_R = p_1 A_1 - p_2 A_2 - \rho Q (V_2 - V_1)$$

\* Note sign convention

# FORCES IN FREE JETS

## MOMENTUM EQUATION APPLIED TO BLADES

### ANGLED BLADE



Use Previous Equation for Flow in Pressure Conduit

#### Reaction in X - Direction

$$F_{RX} = p_1 A_1 - p_2 A_2 \cos\theta - \rho Q (V_2 \cos\theta - V_1)$$

Since  $p_1$  and  $p_2$  are zero

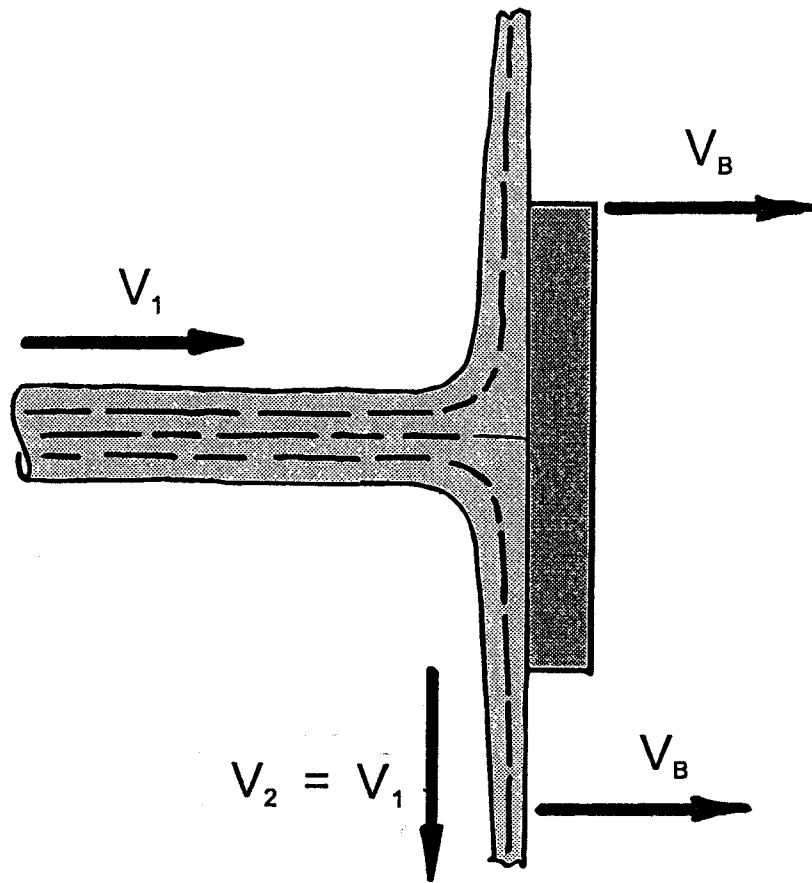
$$F_{RX} = -\rho Q (V_2 \cos\theta - V_1)$$

#### Reaction in Y - Direction

$$F_{RY} = p_2 A_2 \sin\theta + \rho Q V_2 \sin\theta$$

Since  $p_1$  and  $p_2$  are zero

$$F_{RY} = \rho Q V_2 \sin\theta$$



**Figure 1 Jet impinging upon a flat moving plate**

# RESULTANT FORCE

$$F = ma$$

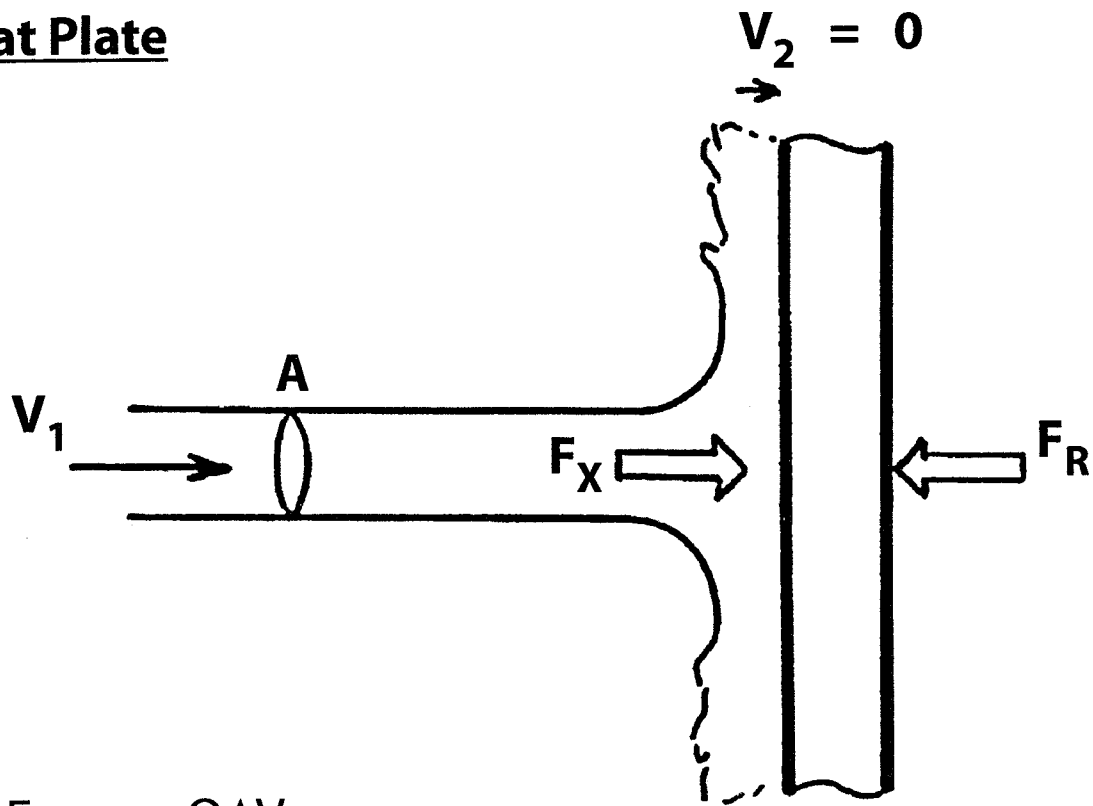
$$F = \rho Q \Delta V$$

$$F_x = \rho Q \Delta V_x$$

$$F_y = \rho Q \Delta V_y$$

$$F_z = \rho Q \Delta V_z$$

## Flat Plate



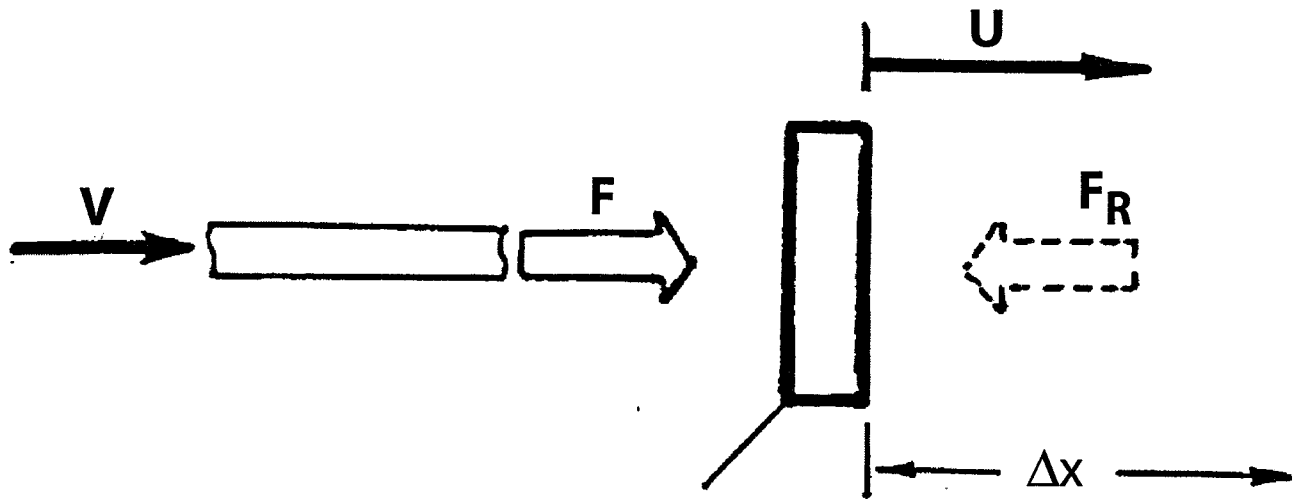
$$F_x = \rho Q \Delta V_x$$

$$= \rho Q (V_1 - V_2)$$

$$= \rho Q (V_1)$$

$$F_R = F_x$$

# WORK AND POWER



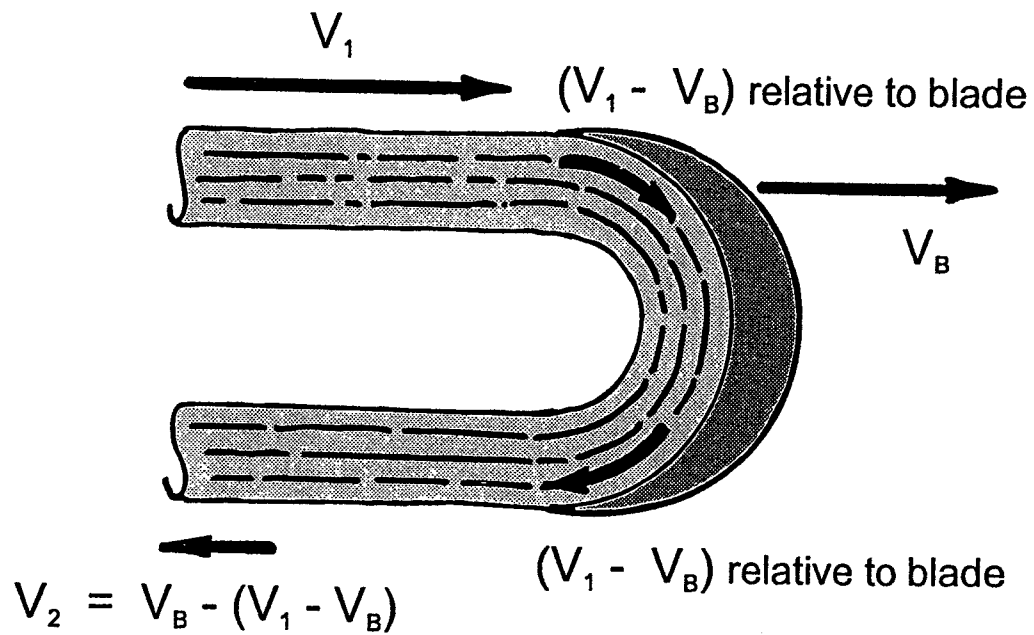
**Force on blade**  $F = \rho Q \Delta V$  (N)

**Work on blade**  $W = \text{Force} \times \text{Distance}$   
 $= F \Delta x$   
 $= \rho Q \Delta V \Delta x$  (Nm)

**Power to blade**  $P = \text{Work} / \text{Time}$   
 $= F \Delta x / \Delta t$   
 $= F U$   
 $= \rho Q \Delta V U$  (J / s)

Now since  $\Delta V = (V - U)$

**Power to blade**  $P = \rho Q (V - U) U$



**Figure 2 Jet impinging upon a curved moving plate**

# FIRST LAW OF THERMODYNAMICS

## (CONSERVATION OF ENERGY)

Energy can neither be created nor destroyed  
but only converted from one form to another.

Therefore

$$dE = \delta Q - \delta W$$

where

$dE$  = small change in  $E$

$\delta Q$  = small amount of  $Q$

$\delta W$  = small amount of  $W$

For a system absorbing heat and doing work

$$dE = \delta Q - \delta W$$

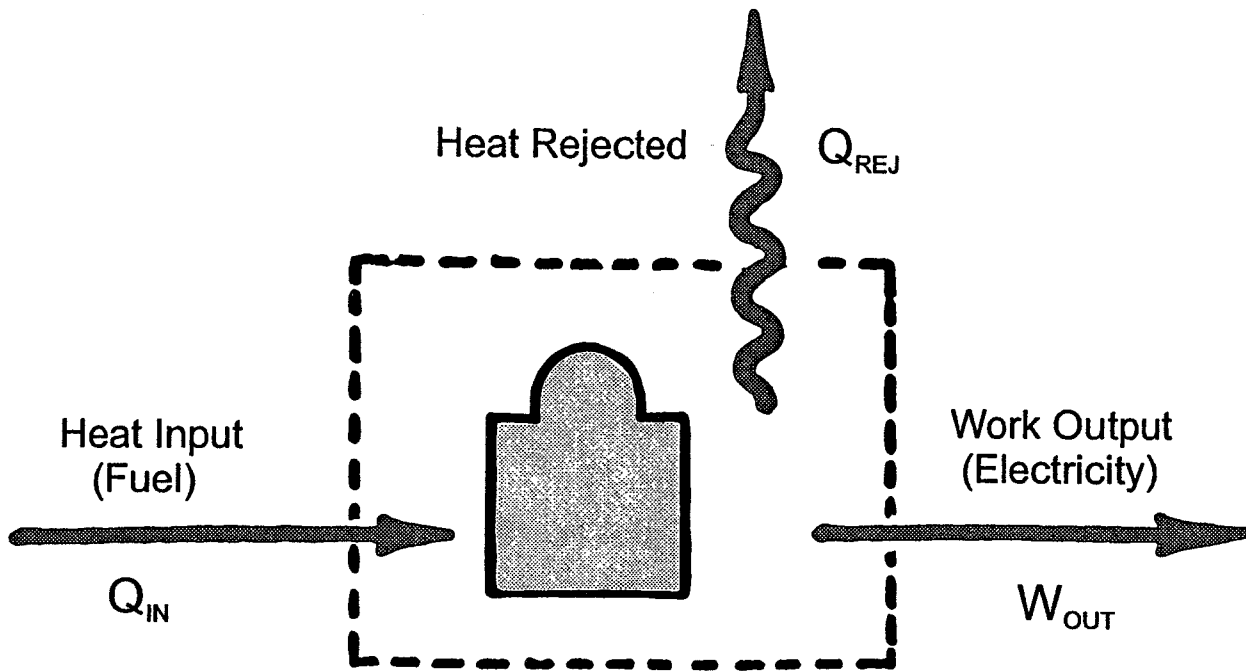
# **SECOND LAW OF THERMODYNAMICS**

**(CONVERSION OF HEAT TO WORK)**

**No heat engine can generate work without net rejection of heat to a low-temperature reservoir.**

**If the cold reservoir could be at absolute zero (no molecular motion and hence no internal energy) then the efficiency of the carnot cycle would be 100% and all the heat would be converted into work.**

**The second law thus defines a portion of heat which is *unavailable* for conversion into work under ambient conditions.**



**Figure 3 Heat and work flow in a heat engine**

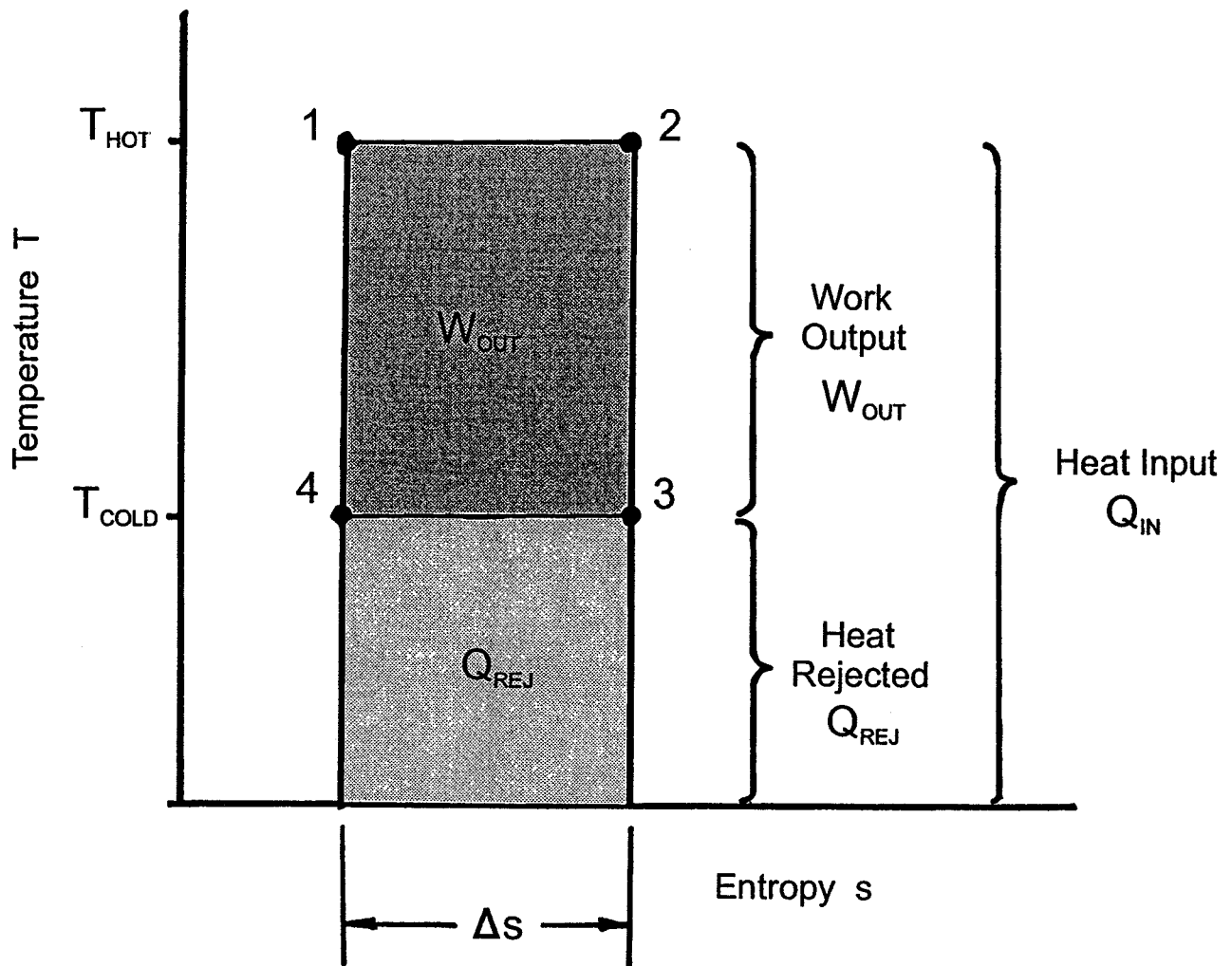
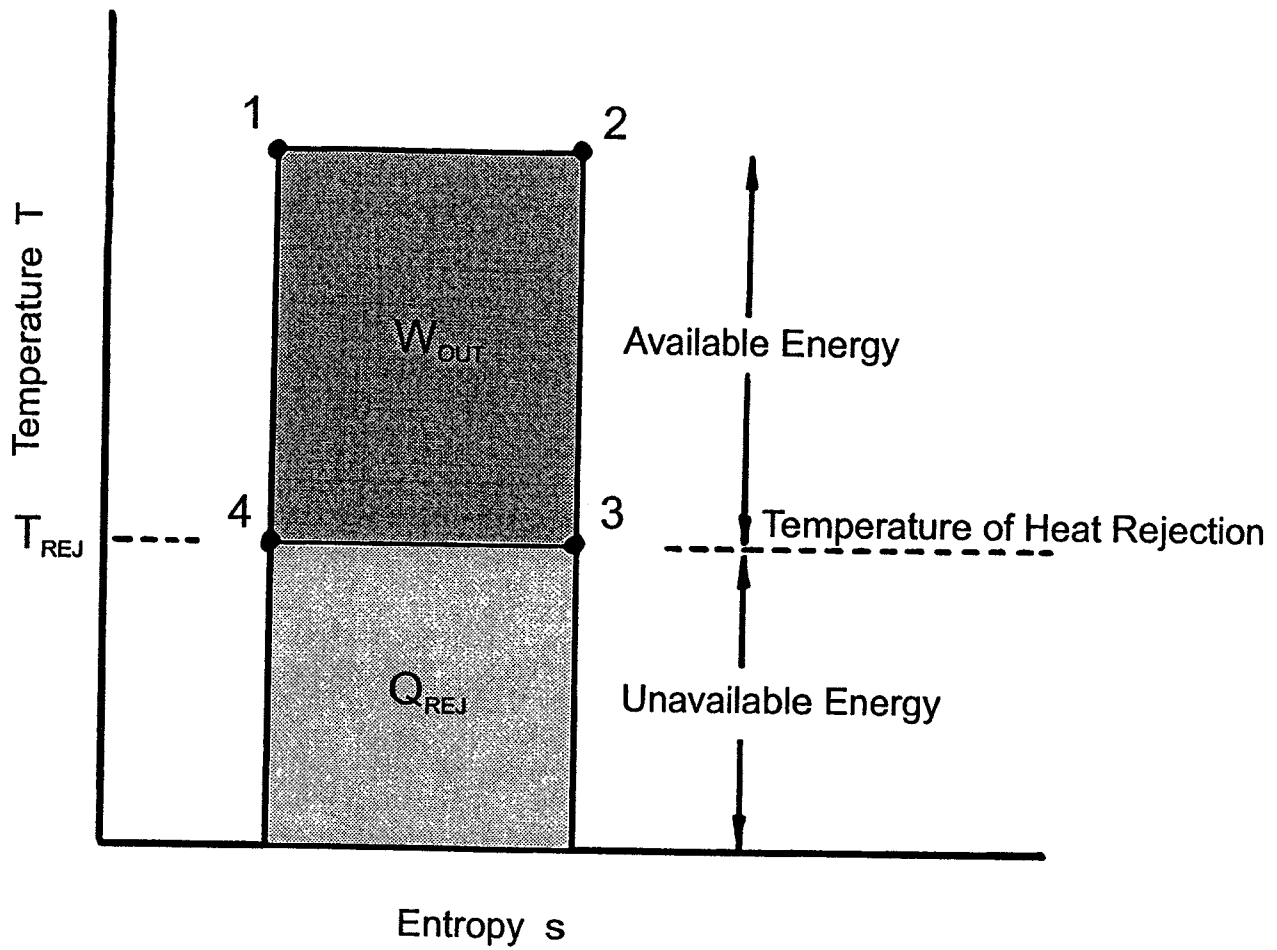
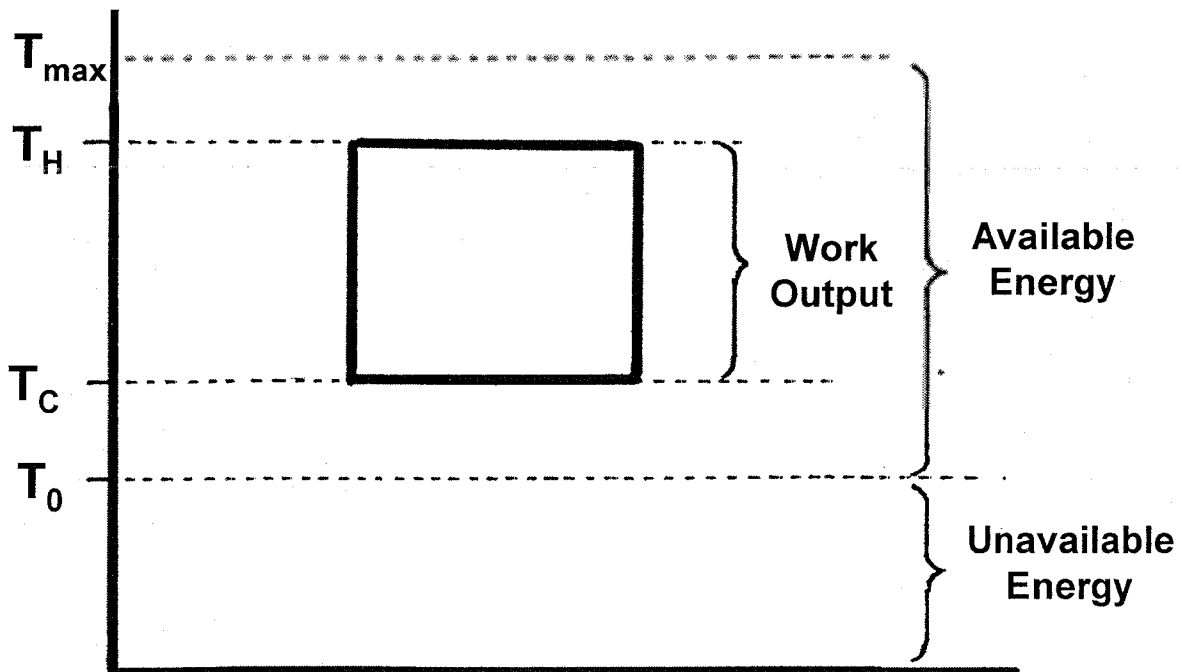
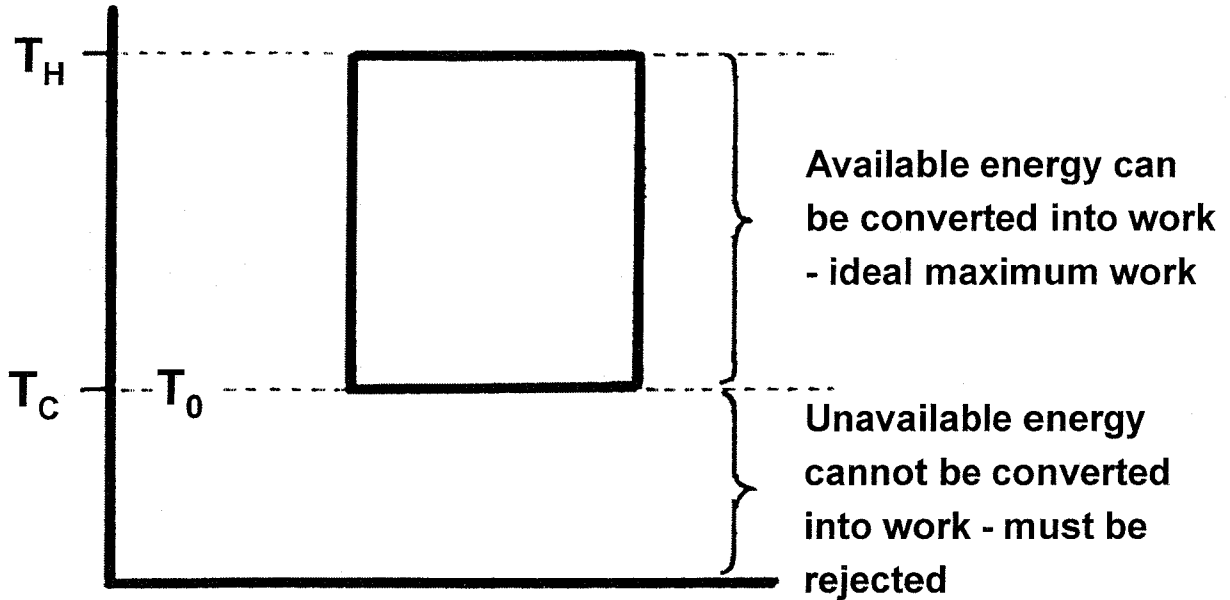


Figure 4 Carnot cycle



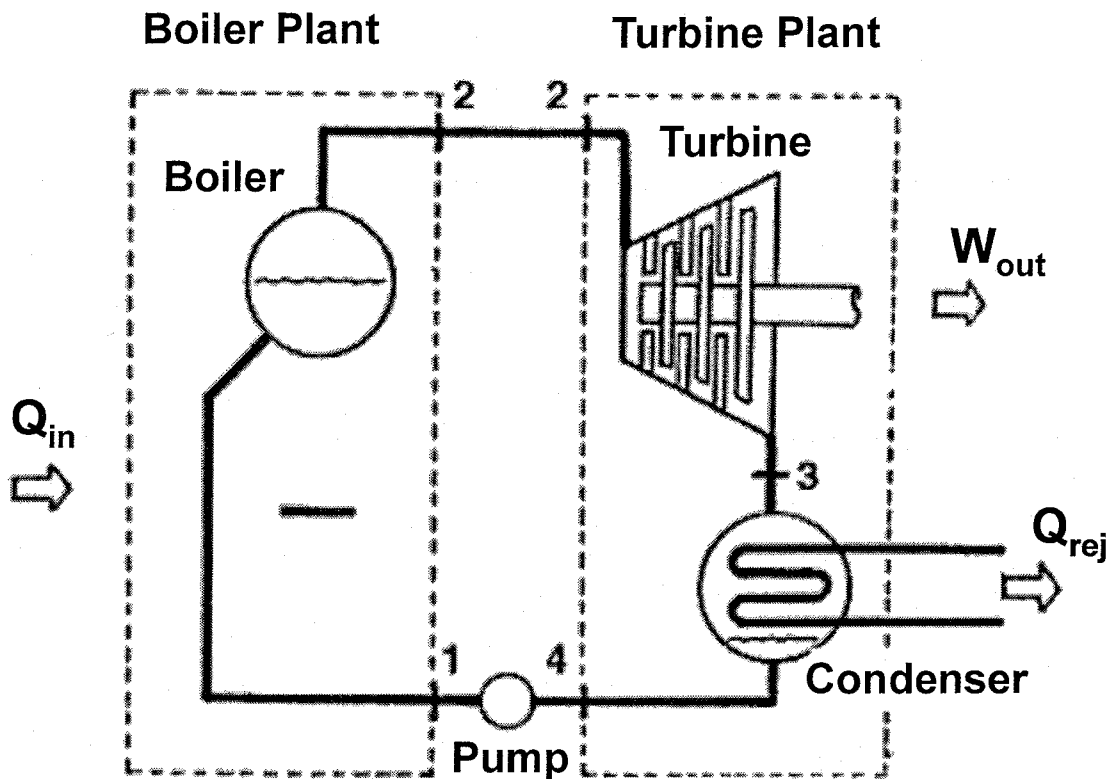
**Figure 5 Available and unavailable energy**

# AVAILABLE AND UNAVAILABLE ENERGY



- $T_0$  = Ambient Temperature (Heat Sink)
- $T_{max}$  = Maximum Temperature of Heat source
- $T_H$  &  $T_C$  = Cycle Working Fluid Temperatures

# STEAM BOILER AND TURBINE



## Component

## Process

Boiler

1 - 2

Turbine

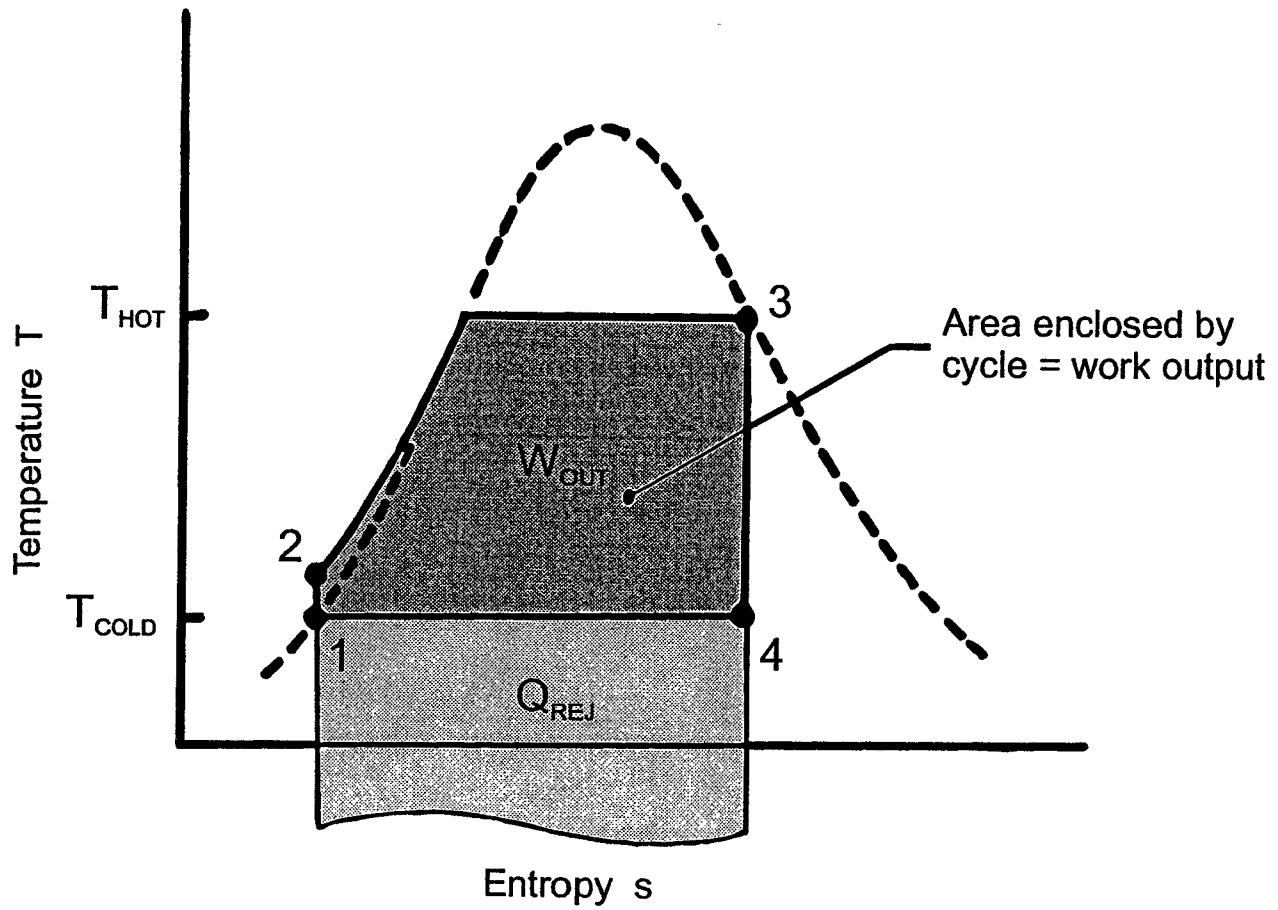
2 - 3

Condenser

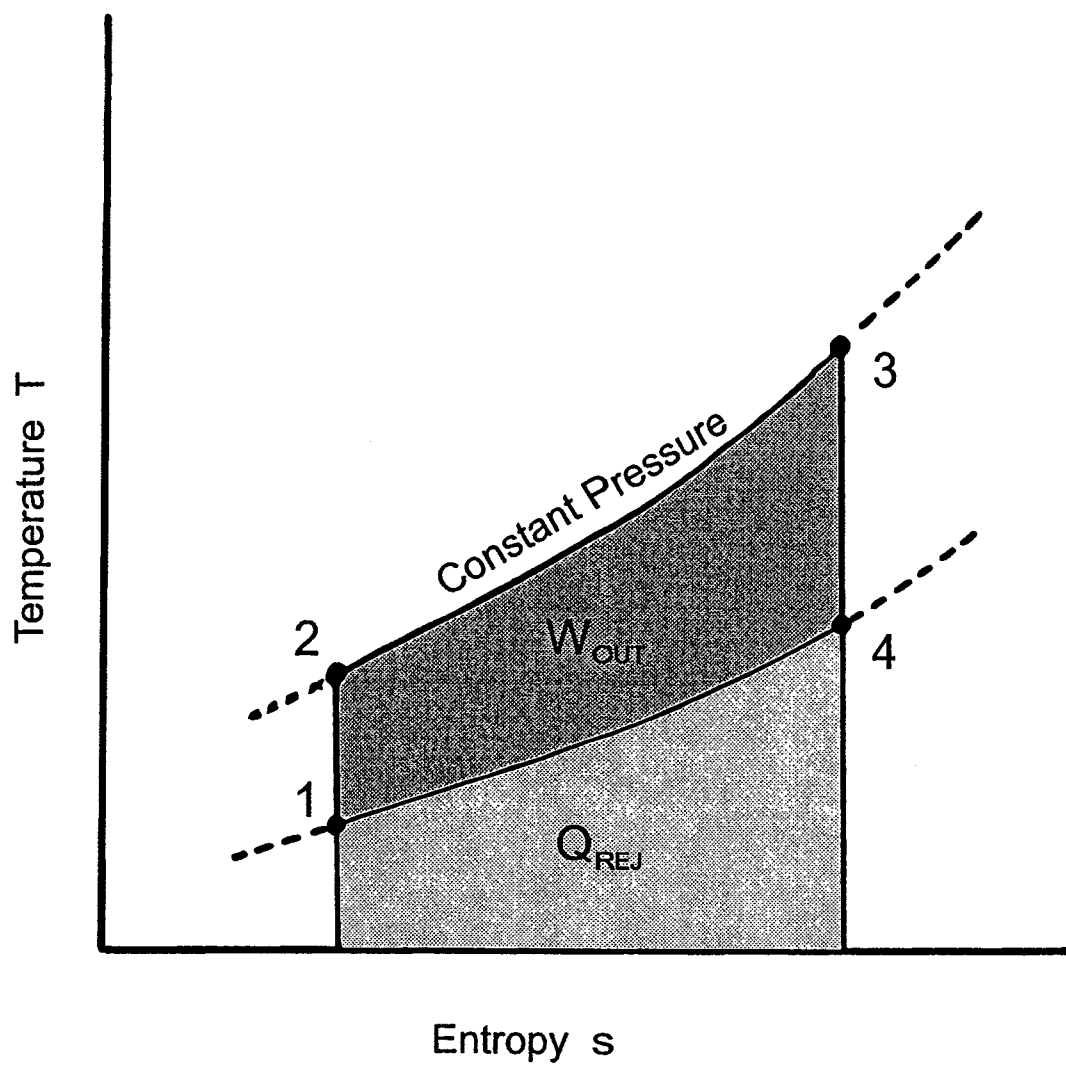
3 - 4

Pump

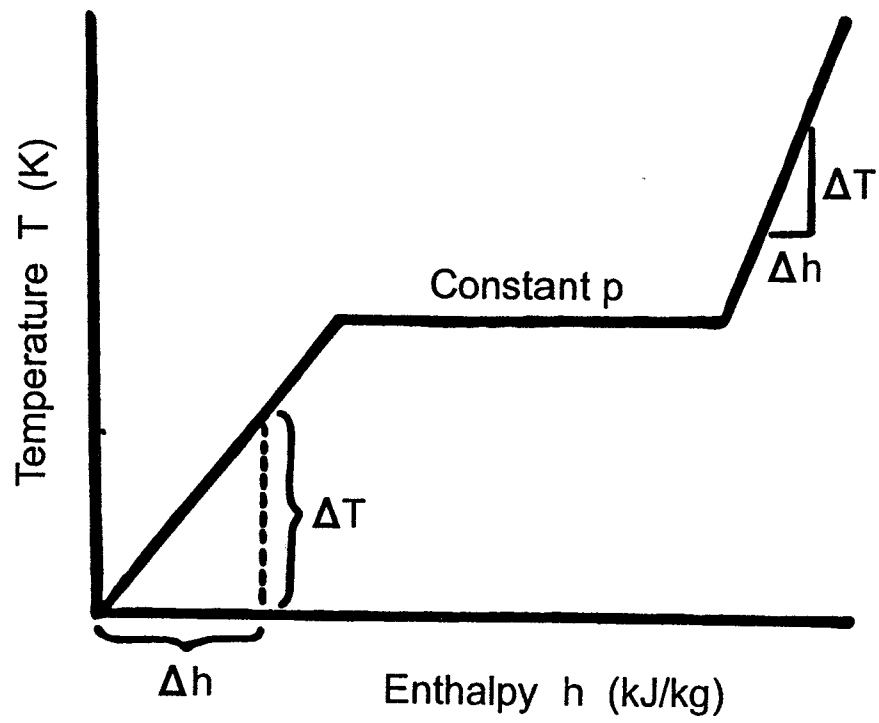
4 - 1



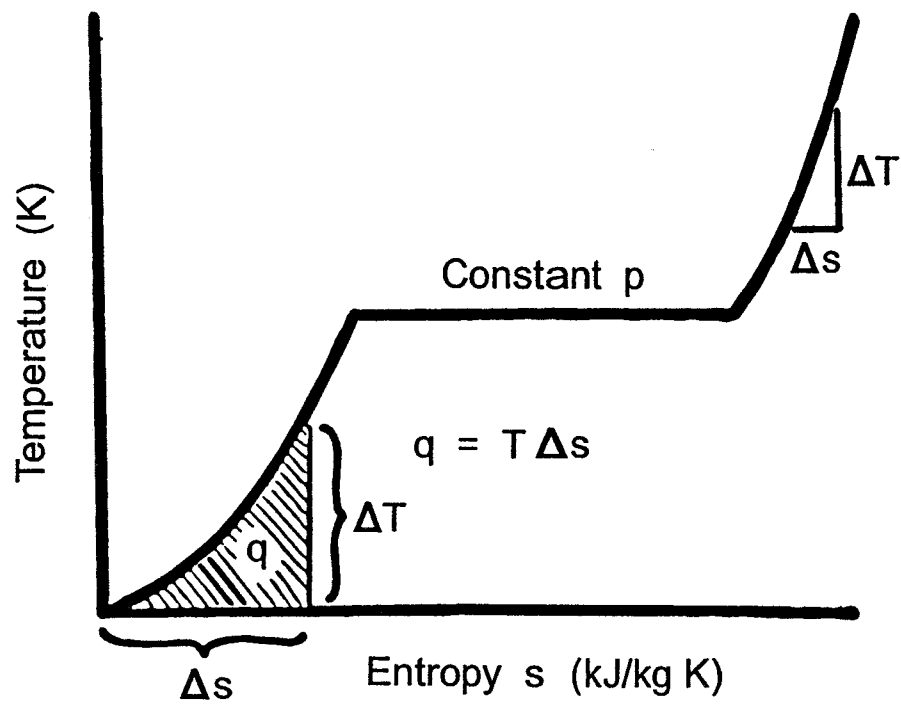
**Figure 6 Rankine cycle for steam turbine**



**Figure 7 Brayton cycle for gas turbines**



**Figure 8 Temperature-enthalpy diagram**



**Figure 9 Temperature-entropy diagram**

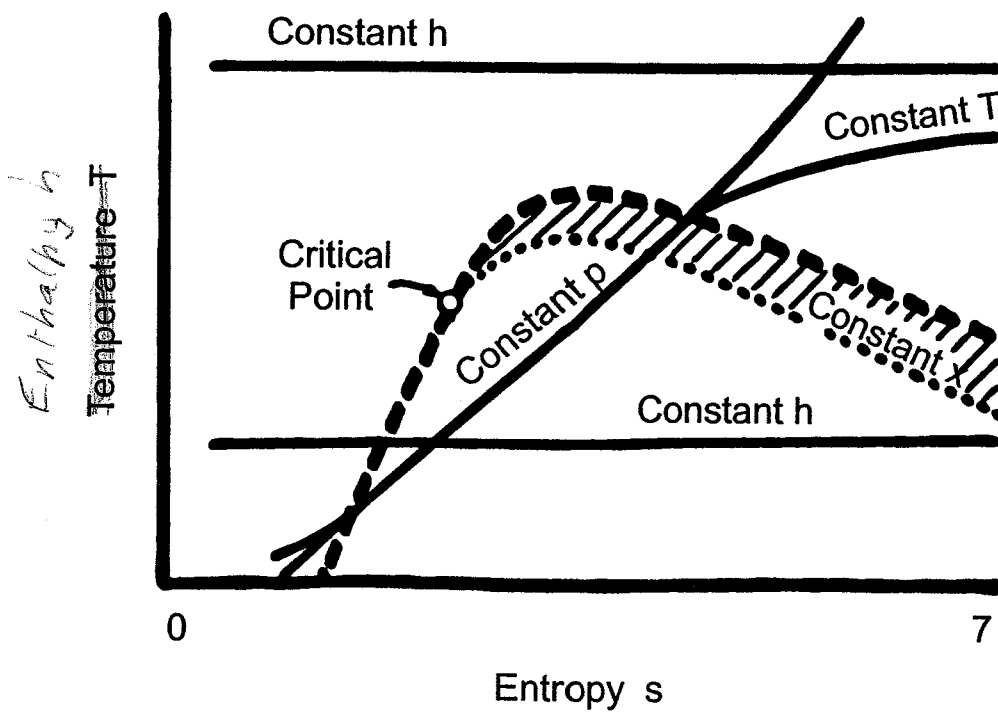
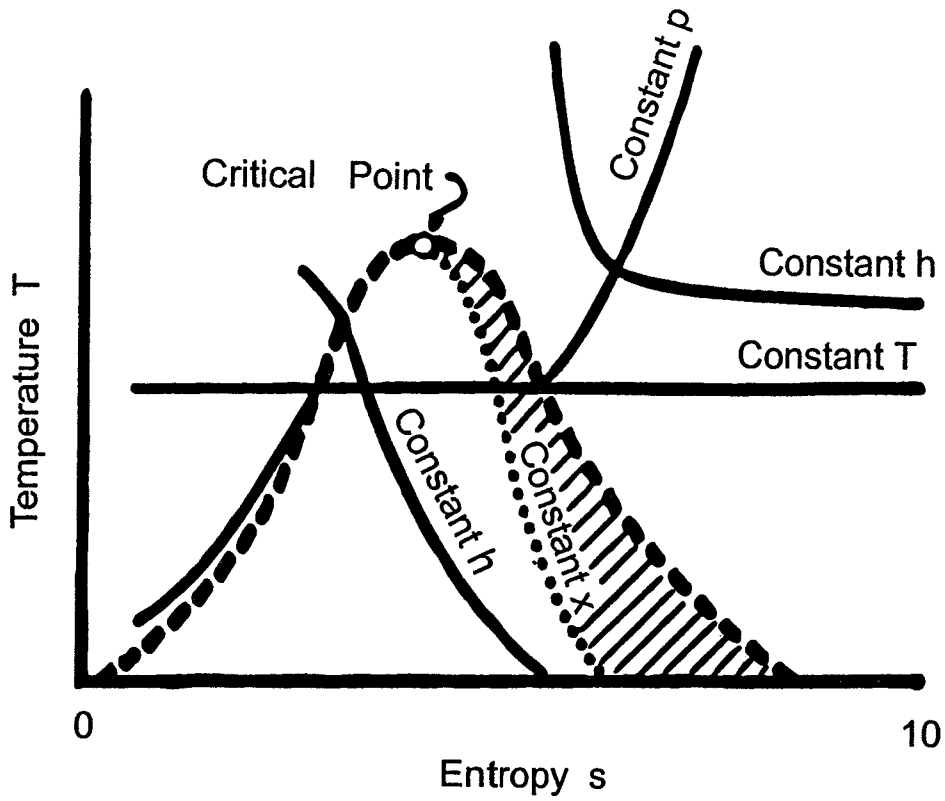


Figure 10. Temperature-entropy and enthalpy-entropy diagrams