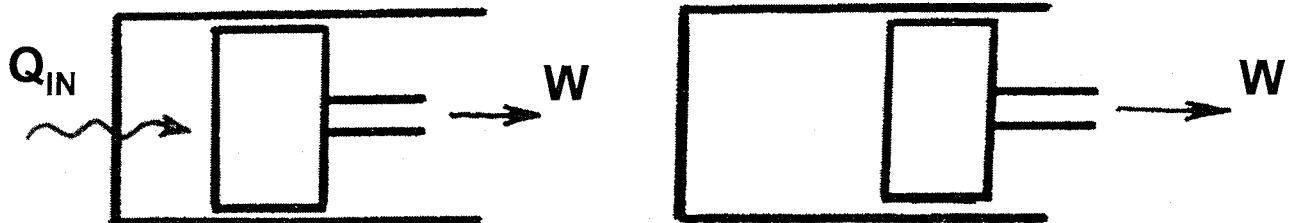


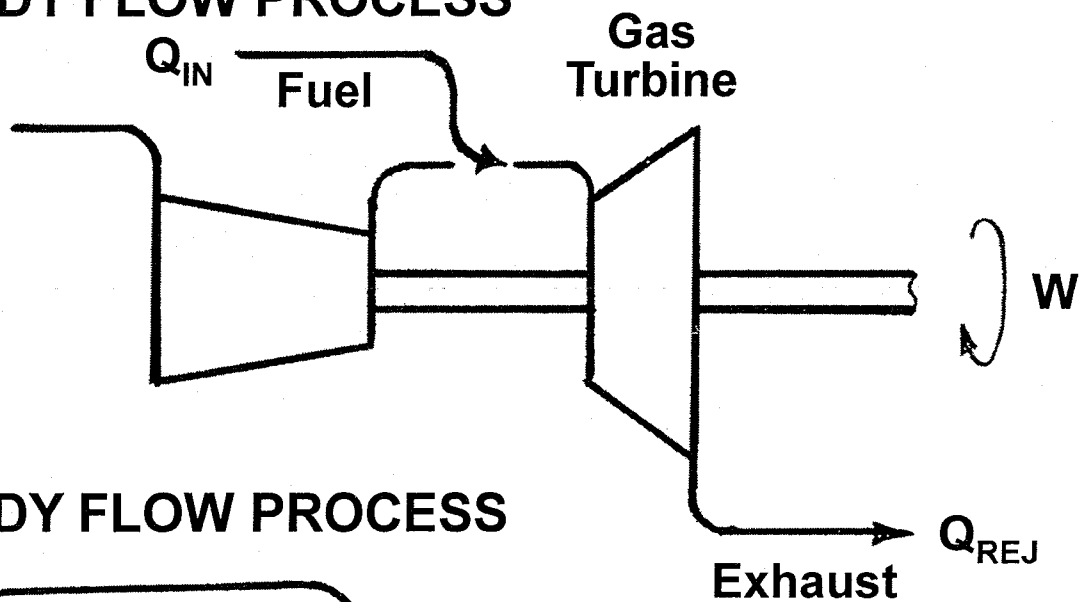
## **STEADY FLOW PROCESSES**

# NON-FLOW AND STEADY FLOW

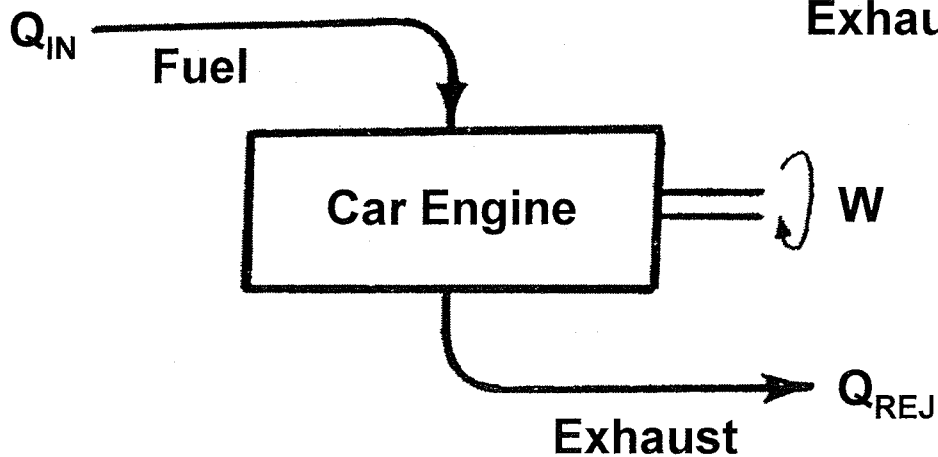
## NON-FLOW PROCESS



## STEADY FLOW PROCESS

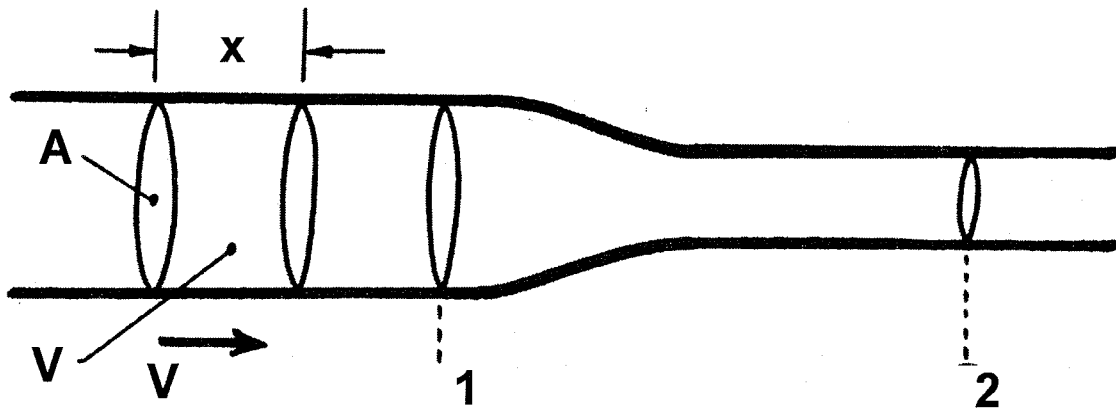


## STEADY FLOW PROCESS



# CONTINUITY EQUATION

(WHAT GOES IN..... MUST COME OUT)



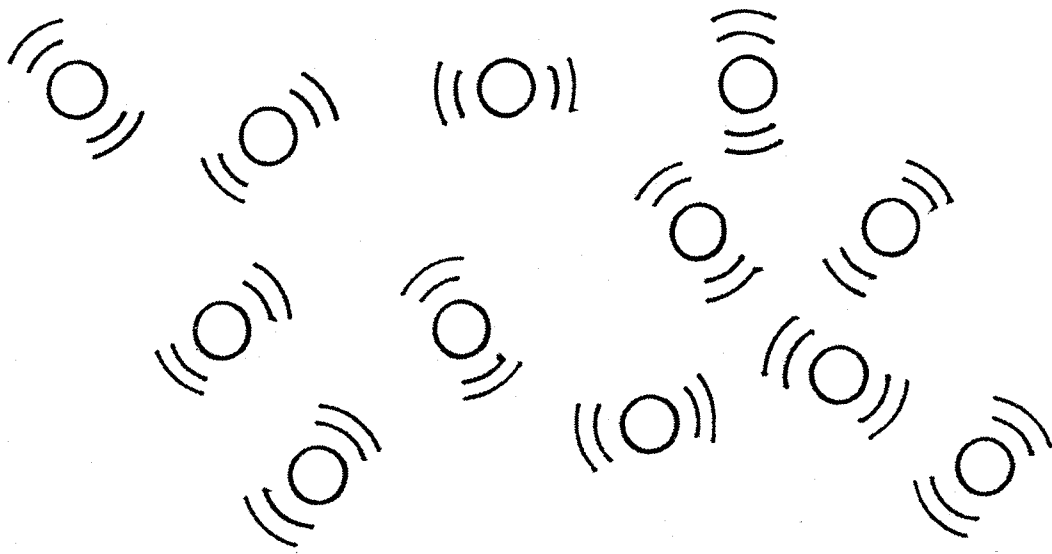
Volume	$V = A x$	$(m^3)$
Density	$\rho$	$(kg/m^3)$
Mass	$m = \rho V = \rho A x$	$(kg)$
Velocity	$V = x / t$	$(m/s)$
	$x = V t$	
Mass	$m = \rho A V t$	$(kg)$
Mass Flow	$M = \rho A V$	$(kg/s)$

Continuity Equation states that:

$$\rho_1 V_1 A_1 = \rho_2 V_2 A_2$$

for steady mass flow rate of  $M$

# INTERNAL ENERGY

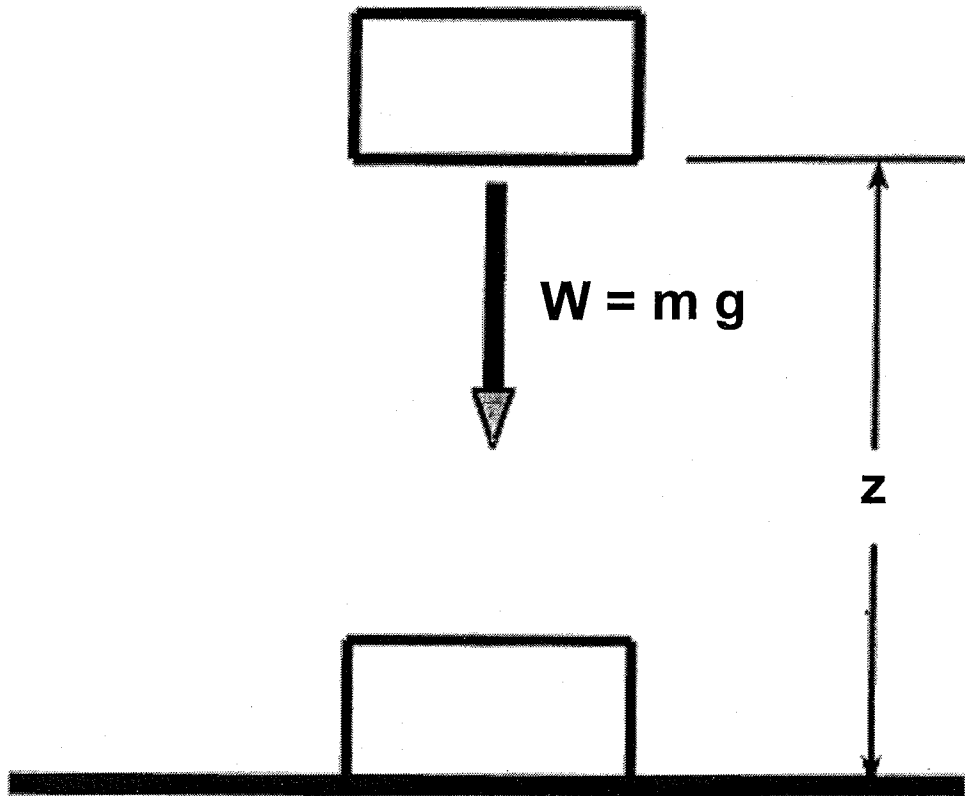


**Internal Energy is due to the motion of the molecules and internal attractive and repulsive forces.**

# POTENTIAL ENERGY

Mass  $m$  at elevation  $z$  above datum plane

Energy = work in lifting mass above datum

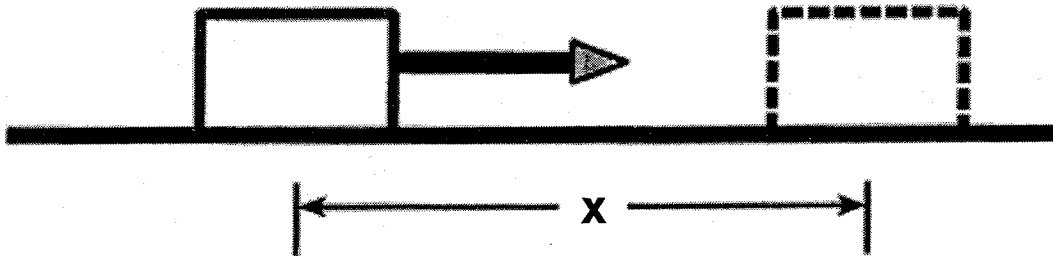


$$\begin{aligned} \text{Potential Energy} &= \text{mass} \times \text{elevation} \times g \\ &= m g z \end{aligned}$$

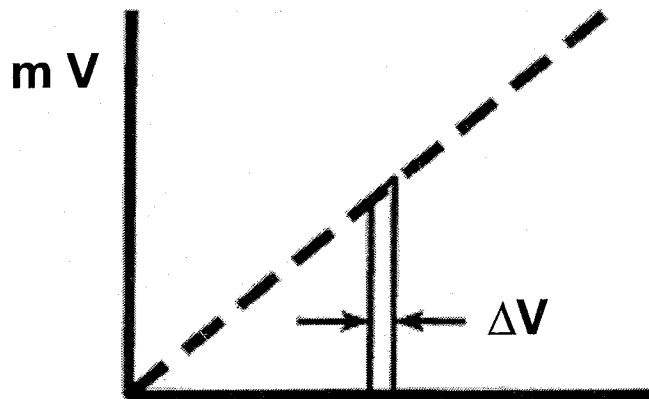
# KINETIC ENERGY

**Kinetic Energy of mass  $m$  at Velocity  $V$**

$$V = x / \Delta t \quad a = \Delta V / \Delta t \quad \Delta t = \Delta V / a$$



**Substituting:  $V = x a / \Delta V$**



$$V \Delta V = a x$$

$$m V \Delta V = m a x$$

$$F x = m V \Delta V$$

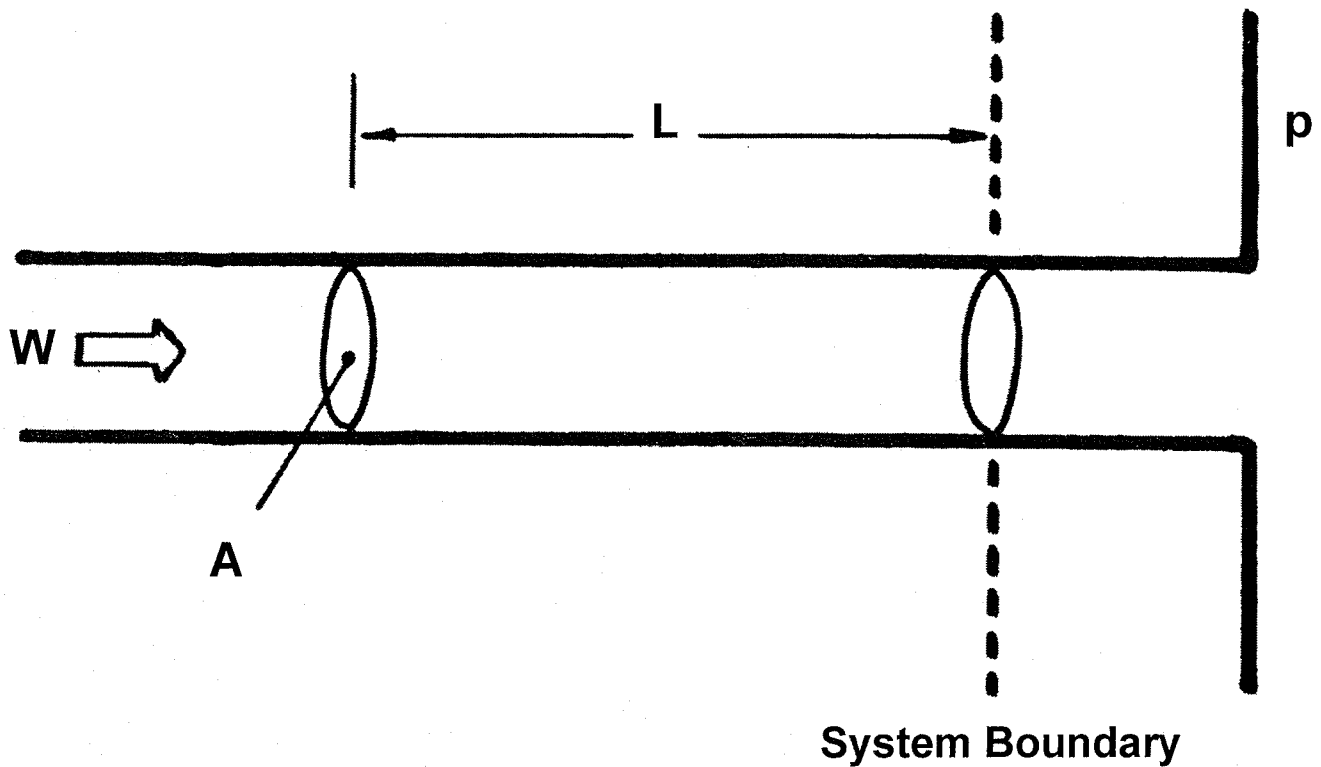
$$\text{Work} = m V \Delta V$$

$$= \frac{1}{2} m V^2$$

**Kinetic Energy =  $\frac{1}{2}$  mass x velocity<sup>2</sup>**

$$= \frac{1}{2} m V^2$$

# FLOW WORK



Work to push plug of fluid through System Boundary is:

$$\begin{aligned} W &= \text{force} \times \text{distance} && (\text{Nm}) \\ &= p A L \\ &= p V && (\text{Nm}) \\ w &= p v && (\text{Nm/kg}) \end{aligned}$$

Flow work:

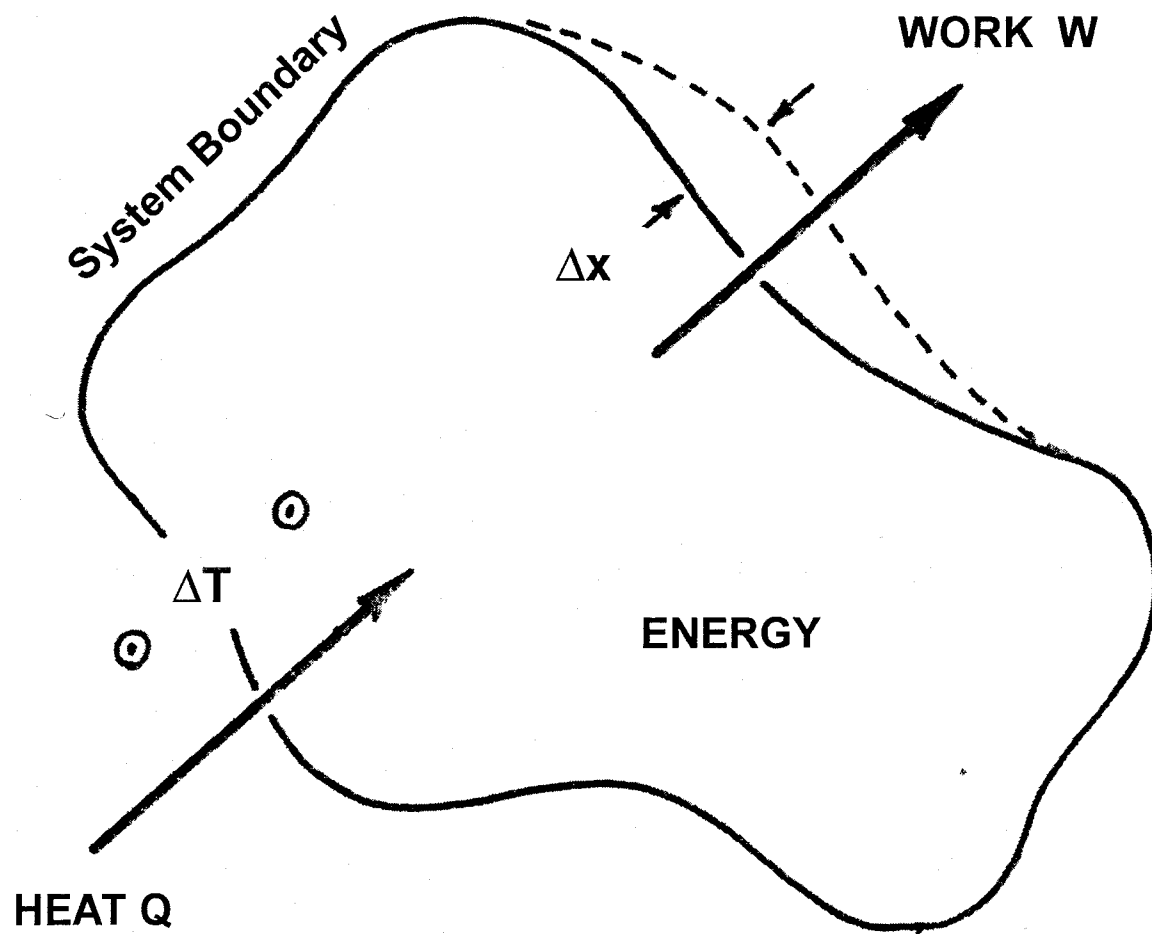
$$w = p v \quad (\text{J/kg})$$

Generally:

$$w = \Delta (pv)$$

# THERMODYNAMIC RELATIONS

## HEAT - ENERGY - WORK



Change in Internal Energy = Heat Input - Work Output

$$\Delta U = Q - W$$

# FIRST LAW OF THERMODYNAMICS

## (CONSERVATION OF ENERGY)

Energy can neither be created nor destroyed  
but only converted from one form to another.

Therefore

$$dE = \delta Q - \delta W$$

where

$dE$  = small change in  $E$

$\delta Q$  = small amount of  $Q$

$\delta W$  = small amount of  $W$

For a system absorbing heat and doing work

$$dE = \delta Q - \delta W$$

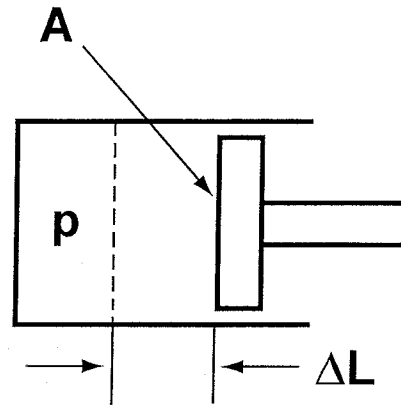
# ENTHALPY DEFINITION

From First Law

$$dU = \delta Q - \delta W$$

$$U_2 - U_1 = Q - W$$

$$u_2 - u_1 = q - w$$



Consider piston in cylinder

$$W = \text{Force} \times \text{Distance}$$

$$= F \times \Delta L$$

$$= pA\Delta L$$

$$= p\Delta V$$

$$w = p\Delta v$$

Substituting and considering  $p$  to be constant

$$u_2 - u_1 = q - p\Delta V$$

$$= q - p(v_2 - v_1)$$

$$q = u_2 - u_1 + p_2v_2 - p_1v_1$$

$$= (u_2 + p_2v_2) - (u_1 + p_1v_1)$$

$$= h_2 - h_1 = \Delta h$$

where  $h = u + pv$  is Enthalpy

# SPECIFIC HEAT

$$\text{Specific Heat} = \frac{\text{Heat Transferred / Unit Mass}}{\text{Change in Temperature}} = \frac{\text{J / kg}}{^{\circ}\text{C}}$$

## Specific heat at constant pressure

$$c_p = \left( \frac{q}{\Delta T} \right)_p$$

$$q = h_2 - h_1 \text{ at constant pressure}$$

$$c_p = \left( \frac{\Delta h}{\Delta T} \right)_p$$

$$q = \Delta h = c_p \Delta T$$

## Specific heat at constant volume

$$c_v = \left( \frac{q}{\Delta T} \right)_v$$

$$q = u_2 - u_1 \text{ at constant volume}$$

$$c_v = \left( \frac{\Delta u}{\Delta T} \right)_v$$

$$q = \Delta u = c_v \Delta T$$

Which is greater :  $c_p$  or  $c_v$ ?

$$k = \frac{c_p}{c_v}$$

**TABLE 2.2**  
**ENERGY BALANCE**

	ENERGY IN (Btu/lb)	ENERGY IN (SI)	ENERGY OUT (Btu/lb)	ENERGY OUT (SI)
Potential energy	$\frac{Z_1 g}{J g_c}$	$Z_1 g$	$\frac{Z_2 g}{J g_c}$	$Z_2 g$
Kinetic energy	$\frac{V_1^2}{2 g_c J}$	$\frac{V_1^2}{2}$	$\frac{V_2^2}{2 g_c J}$	$\frac{V_2^2}{2}$
Internal energy	$u_1$	$u_1$	$u_2$	$u_2$
Flow work	$\frac{p_1 v_1}{J}$	$p_1 v_1$	$\frac{p_2 v_2}{J}$	$p_2 v_2$
Work	$\frac{W_{in}}{J}$	$W_{in}$	$\frac{W_{out}}{J}$	$W_{out}$
Heat	$q_{in}$	$q_{in}$	$q_{out}$	$q_{out}$

# ENERGY EQUATION

THE FIRST LAW OF THERMODYNAMICS STATES THAT ENERGY IS NEITHER CREATED NOR DESTROYED...

Combining all types of energy into a single equation gives:

$$\begin{aligned} z_1 g + V_1^2 / 2 + u_1 + p_1 v_1 + w_{IN} + q_{IN} \\ = z_2 g + V_2^2 / 2 + u_2 + p_2 v_2 + w_{OUT} + q_{OUT} \end{aligned}$$

For constant  $u$  and no  $w$  or  $q$  this equation reduces to the Bernoulli Equation

$$z_1 g + V_1^2 / 2 + p_1 v_1 = z_2 g + V_2^2 / 2 + p_2 v_2$$

$$z_1 + V_1^2 / 2g + p_1 / \rho g = z_2 + V_2^2 / 2g + p_2 / \rho g$$

Note that:  $u + pv = h$

Hence rewriting gives:

$$\begin{aligned} (z_1 - z_2)g + (V_1^2 - V_2^2) / 2 + (h_1 - h_2) + (w_{IN} - w_{OUT}) \\ + (q_{IN} - q_{OUT}) = 0 \end{aligned}$$

This equation can be adapted to all flow processes by substituting appropriate values.

# TURBINE

## ENERGY EQUATION

$$z_1 g + V_1^2/2 + u_1 + p_1 v_1 + w_{IN} + q_{IN} \\ = z_2 g + V_2^2/2 + u_2 + p_2 v_2 + w_{OUT} + q_{OUT}$$

## ASSUMPTIONS

No change in elevation

$$z_2 = z_1$$

Small change in velocity

$$V_2 = V_1$$

No work input

$$w_{IN} = 0$$

No heat input

$$q_{IN} = 0$$

No heat output

$$q_{OUT} = 0$$

## TURBINE EQUATION

$$u_1 + p_1 v_1 = u_2 + p_2 v_2 + w_{OUT} \\ h_1 = h_2 + w_{OUT}$$

$$w_{OUT} = h_1 - h_2$$

Generally

$$w_{OUT} = \Delta h$$

## FOR GAS TURBINE

$$w_{OUT} = c_p \Delta T$$

# BOILER

## ENERGY EQUATION (STEAM SIDE)

$$z_1g + V_1^2/2 + u_1 + p_1v_1 + w_{IN} + q_{IN} \\ = z_2g + V_2^2/2 + u_2 + p_2v_2 + w_{OUT} + q_{OUT}$$

## ASSUMPTIONS

No change in elevation

Small change in velocity

No work input

No work output

No heat output from steam

$$z_2 = z_1$$

$$V_2 = V_1$$

$$w_{IN} = 0$$

$$w_{OUT} = 0$$

$$q_{OUT} = 0$$

## BOILER EQUATION

$$u_1 + p_1v_1 + q_{IN} = u_2 + p_2v_2$$

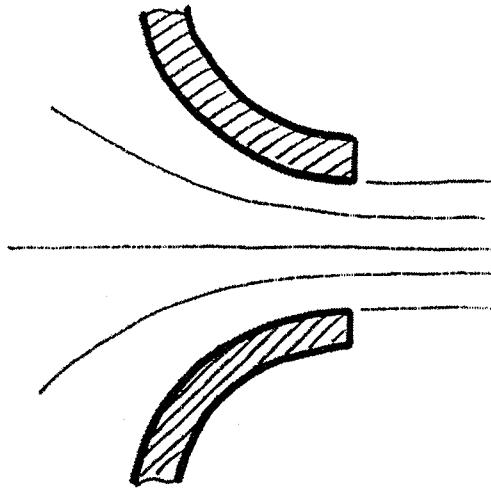
$$h_1 + q_{IN} = h_2$$

$$q_{IN} = h_2 - h_1$$

Generally

$$\text{Heat in} = \Delta h$$

# NOZZLE AND THROTTLE



## NOZZLE

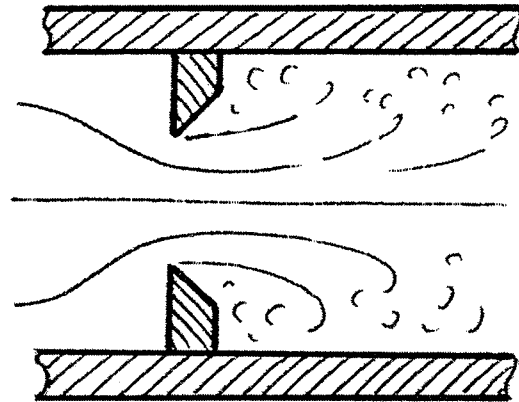
No work

No heat transfer

No friction

Reversible adiabatic  
(isentropic)

**PURPOSE:**  
To produce high  
velocity flow



## THROTTLE

No work

No heat transfer

Friction due to turbulence

Irreversible adiabatic  
(isenthalpic)

**PURPOSE:**  
To reduce pressure  
or restrict flow

# NOZZLE

## ENERGY EQUATION

$$z_1g + V_1^2/2 + u_1 + p_1v_1 + w_{IN} + q_{IN} \\ = z_2g + V_2^2/2 + u_2 + p_2v_2 + w_{OUT} + q_{OUT}$$

## ASSUMPTIONS

No change in elevation

$$z_2 = z_1$$

No work input

$$w_{IN} = 0$$

No work output

$$w_{OUT} = 0$$

No heat input

$$q_{IN} = 0$$

No heat output

$$q_{OUT} = 0$$

## NOZZLE EQUATION

$$V_1^2/2 + u_1 + p_1v_1 = V_2^2/2 + u_2 + p_2v_2$$

$$V_1^2/2 + h_1 = V_2^2/2 + h_2$$

$$V_2^2 - V_1^2 = 2(h_1 - h_2)$$

If initial velocity is negligible

$$V_2 = \sqrt{2(h_1 - h_2)}$$

For gas flow:  $V_2 = \sqrt{2c_p\Delta T}$

Generally: Velocity = Function of  $\sqrt{\Delta h}$

# THROTTLE

## ENERGY EQUATION

$$z_1g + V_1^2/2 + u_1 + p_1v_1 + w_{IN} + q_{IN} \\ = z_2g + V_2^2/2 + u_2 + p_2v_2 + w_{OUT} + q_{OUT}$$

## ASSUMPTIONS

No change in elevation

Small change in velocity

No work input

No work output

No heat input

No heat output

$$z_2 = z_1$$

$$V_2 = V_1$$

$$w_{IN} = 0$$

$$w_{OUT} = 0$$

$$q_{IN} = 0$$

$$q_{OUT} = 0$$

## THROTTLE EQUATION

$$u_1 + p_1v_1 = u_2 + p_2v_2$$

$$h_1 = h_2$$

## CONSTANT ENTHALPY PROCESS

The Throttling Process involves acceleration of the fluid in a constriction (as in a nozzle) followed by turbulent deceleration with frictional heating as the flow velocity returns to normal.

# HYDRO TURBINE PLANT

## ENERGY EQUATION (WHOLE PLANT)

$$\begin{aligned} z_1 g + V_1^2/2 + u_1 + p_1 v_1 + w_{IN} + q_{IN} \\ = z_2 g + V_2^2/2 + u_2 + p_2 v_2 + w_{OUT} + q_{OUT} \end{aligned}$$

## ASSUMPTIONS

Small change in velocity	$V_2 = V_1$
Little change in internal energy	$u_2 = u_1$
No change in pressure	$p_2 = p_1$
Little change in specific volume	$v_2 = v_1$
No work input	$w_{IN} = 0$
No heat input	$q_{IN} = 0$
No heat output	$q_{OUT} = 0$

## HYDRO TURBINE EQUATION

$$\begin{aligned} z_1 g &= z_2 g + w_{OUT} \\ w_{OUT} &= (z_1 - z_2)g \end{aligned}$$

Generally

$$\text{Work out} = g \Delta z$$

Power

$$P = \rho g Q H \quad \text{where } H = \Delta z$$

# PUMP

## (GENERAL EQUATION)

### ENERGY EQUATION (ACROSS PUMP)

$$z_1 g + V_1^2/2 + u_1 + p_1 v_1 + w_{IN} + q_{IN} \\ = z_2 g + V_2^2/2 + u_2 + p_2 v_2 + w_{OUT} + q_{OUT}$$

### ASSUMPTIONS

No change in elevation

Small change in velocity

No work output

No heat input

No heat output

$$z_2 = z_1$$

$$V_2 = V_1$$

$$w_{OUT} = 0$$

$$q_{IN} = 0$$

$$q_{OUT} = 0$$

### PUMP EQUATION

$$u_1 + p_1 v_1 + w_{IN} = u_2 + p_2 v_2$$

$$h_1 + w_{IN} = h_2$$

$$w_{IN} = h_2 - h_1$$

Generally

$$\text{Work in} = \Delta h$$

# COMPRESSOR

## Energy Equation

$$\begin{aligned} z_1 g + V_1^2/2 + u_1 + p_1 v_1 + w_{in} + q_{in} \\ = z_2 g + V_2^2/2 + u_2 + p_2 v_2 + w_{out} + q_{out} \end{aligned}$$

## Assumptions

No change in elevation

Small change in velocity

No work output

No heat input

No heat output

$$z_2 = z_1$$

$$V_2 = V_1$$

$$w_{out} = 0$$

$$q_{in} = 0$$

$$q_{out} = 0$$

## Compressor Equation

$$u_1 + p_1 v_1 + w_{in} = u_2 + p_2 v_2$$

$$h_1 + w_{in} = h_2$$

$$w_{in} = h_2 - h_1$$

$$\text{Work in} = \Delta h$$

$$w_{in} = c_p (T_2 - T_1)$$

# PUMP

## ENERGY EQUATION

$$\begin{aligned} z_1 g + V_1^2/2 + u_1 + p_1 v_1 + w_{IN} + q_{IN} \\ = z_2 g + V_2^2/2 + u_2 + p_2 v_2 + w_{OUT} + q_{OUT} \end{aligned}$$

## ASSUMPTIONS

No change in elevation

$$z_2 = z_1$$

Small change in velocity

$$V_2 = V_1$$

Little change in internal energy

$$u_2 = u_1$$

Little change in specific volume

$$v_2 = v_1 = v$$

No work output

$$w_{OUT} = 0$$

No heat input

$$q_{IN} = 0$$

No heat output

$$q_{OUT} = 0$$

## PUMP EQUATION

$$p_1 v + w_{IN} = p_2 v$$

$$w_{IN} = (p_2 - p_1) v$$

$$w_{IN} = (p_2 - p_1) / \rho$$

Generally

$$\text{Work in} = \Delta p / \rho$$

# HEAT EXCHANGER

## Energy Equation

$$\begin{aligned} z_1 g + V_1^2/2 + u_1 + p_1 v_1 + w_{in} + q_{in} \\ = z_2 g + V_2^2/2 + u_2 + p_2 v_2 + w_{out} + q_{out} \end{aligned}$$

## Assumptions

No change in elevation

$$z_2 = z_1$$

Small change in velocity

$$V_2 = V_1$$

No work input

$$w_{in} = 0$$

No work output

$$w_{out} = 0$$

## Equation for hot fluid

$$u_1 + p_1 v_1 + \cancel{q_{in}} = u_2 + p_2 v_2 + q_{out}$$

$$h_1 - h_2 = q_{out} \quad (\text{J/kg})$$

$$M_{hot} (h_1 - h_2)_{hot} = M_{hot} q_{out} \quad (\text{J/s})$$

## Equation for cold fluid

$$u_1 + p_1 v_1 + q_{in} = u_2 + p_2 v_2 + \cancel{q_{out}}$$

$$q_{in} = h_2 - h_1 \quad (\text{J/kg})$$

$$M_{cold} q_{in} = M_{cold} (h_2 - h_1)_{cold} \quad (\text{J/s})$$

$$M_{hot} (h_1 - h_2)_{hot} = M_{cold} (h_2 - h_1)_{cold}$$

# ENERGY EQUATION APPLICATIONS

## THINGS TO REMEMBER

### ENTHALPY $h$

$$u_1 + p_1 v_1 - u_2 - p_2 v_2 = h_1 - h_2$$

### SPECIFIC HEAT $c_p$ AND $c_v$

$$h_1 - h_2 = c_p (T_1 - T_2)$$

$$u_1 - u_2 = c_v (T_1 - T_2)$$

### UNITS OF ENERGY EQUATION

$$\text{J / kg} \quad (\text{h and u are in kJ/kg})$$

### KINETIC ENERGY $V^2/2$

$$\left(\frac{\text{m}}{\text{s}}\right)^2 = \frac{\text{kg}}{\text{kg}} \times \frac{\text{m m}}{\text{s s}} = \frac{\text{Nm}}{\text{kg}} = \frac{\text{J}}{\text{kg}}$$

### RATE OF HEAT FLOW OR POWER

$$= \text{energy} \times \text{mass flow rate}$$

$$= \frac{\text{J}}{\text{kg}} \times \frac{\text{kg}}{\text{s}} = \frac{\text{J}}{\text{s}} = \text{W}$$