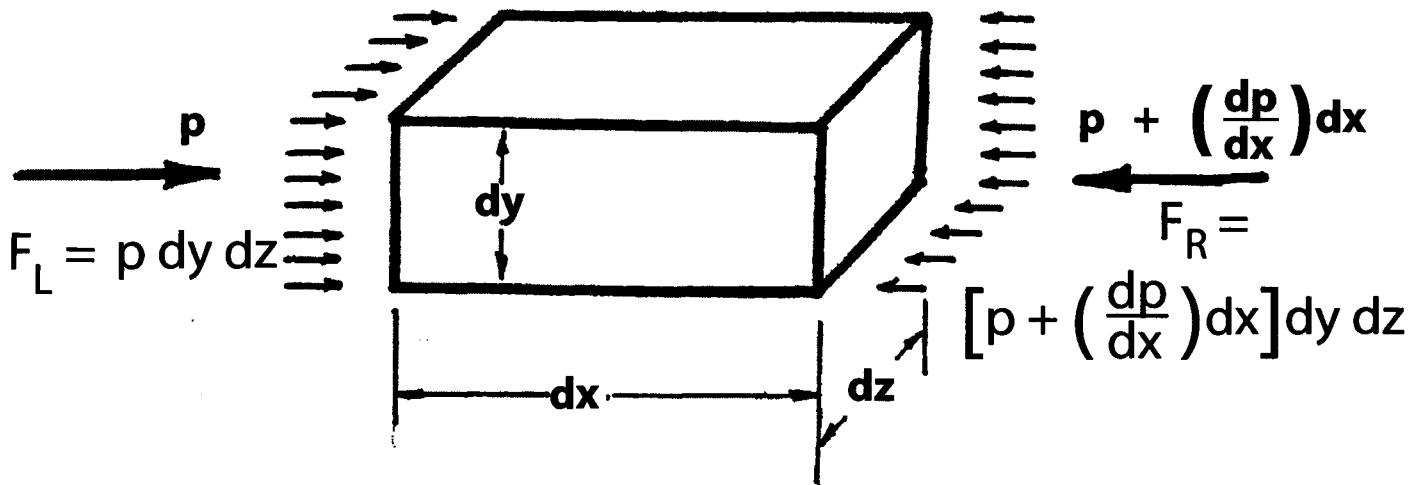


# **LECTURE 0**

## **INTRODUCTION MOMENTUM THEORY**

# EULER'S EQUATION

(FORCE = MASS X ACCELERATION)



**Net Force**

$$\begin{aligned} F &= F_L - F_R \\ &= p \, dy \, dz - \left[ p + \left(\frac{dp}{dx}\right) dx \right] dy \, dz \\ &= -\frac{dp}{dx} (dx \, dy \, dz) \end{aligned}$$

**Fluid Mass**

$$M = \rho (dx \, dy \, dz)$$

**Acceleration**

$$\begin{aligned} a &= \frac{dV}{dt} \\ &= \frac{dV}{dx} \frac{dx}{dt} \\ &= \frac{dV}{dx} v \end{aligned}$$

**Net Force**

$$F = m a$$

$$-\frac{dp}{dx} (dx \, dy \, dz) = \rho (dx \, dy \, dz) \frac{dV}{dx} v$$

$$\boxed{dp = -\rho v \, dV}$$

# **MOMENTUM EQUATION**

## **(CHANGE IN FLOW CREATES FORCE)**

### **Euler's Equation**

$$dp = -\rho V dV$$

Force equals Pressure x Area

$$A dp = -\rho VA dV$$

If  $\rho VA = M$  is constant

$$A dp = -M dV$$

**Integrating between ① and ②**

$$A \int_1^2 dp = -M \int_1^2 dV$$

$$A(p_2 - p_1) = -M(V_2 - V_1)$$

**If there is an external residual force acting in x-direction**

$$A(p_2 - p_1) \neq -M(V_2 - V_1)$$

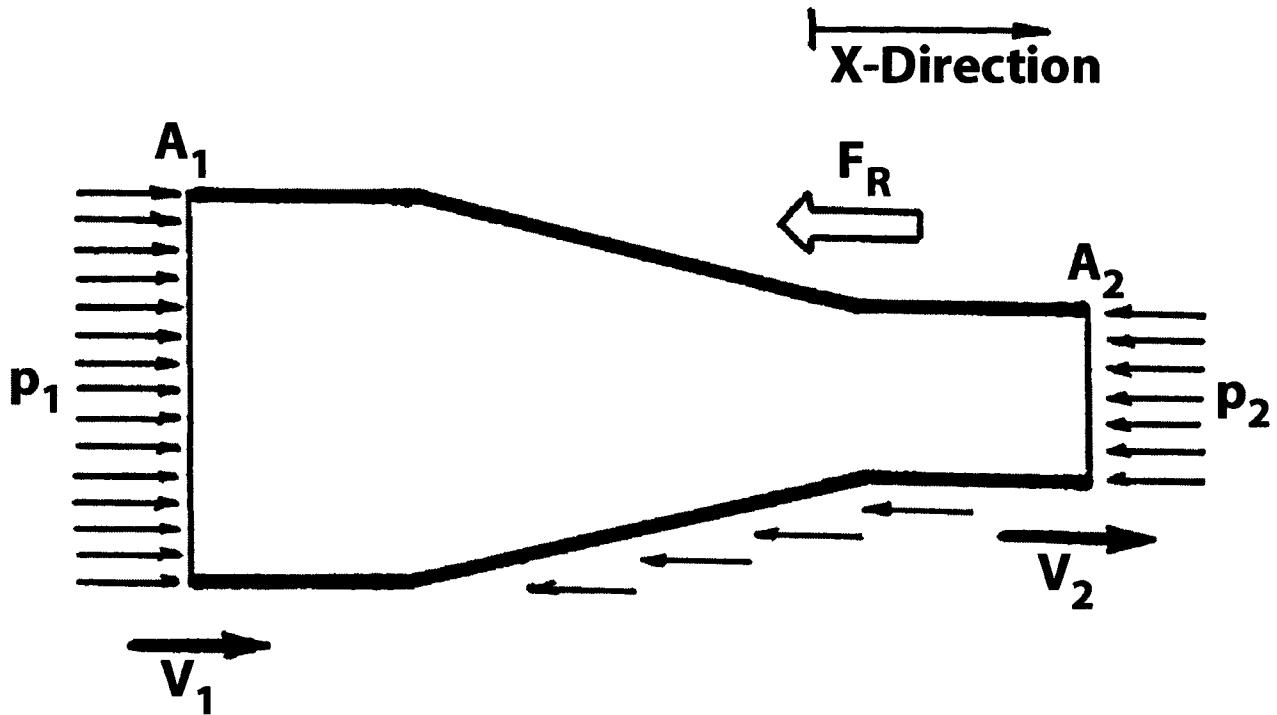
**Force is the difference between the above terms**

$$F = -A(p_2 - p_1) - M(V_2 - V_1)$$

$$F = p_1 A - p_2 A - M(V_2 - V_1)$$

# APPLICATIONS

## CONICAL PIPE



### Forces in X - Direction (on fluid)

$$\Sigma F_x = p_1 A_1 - p_2 A_2 - F_R = \rho Q \Delta V_x$$

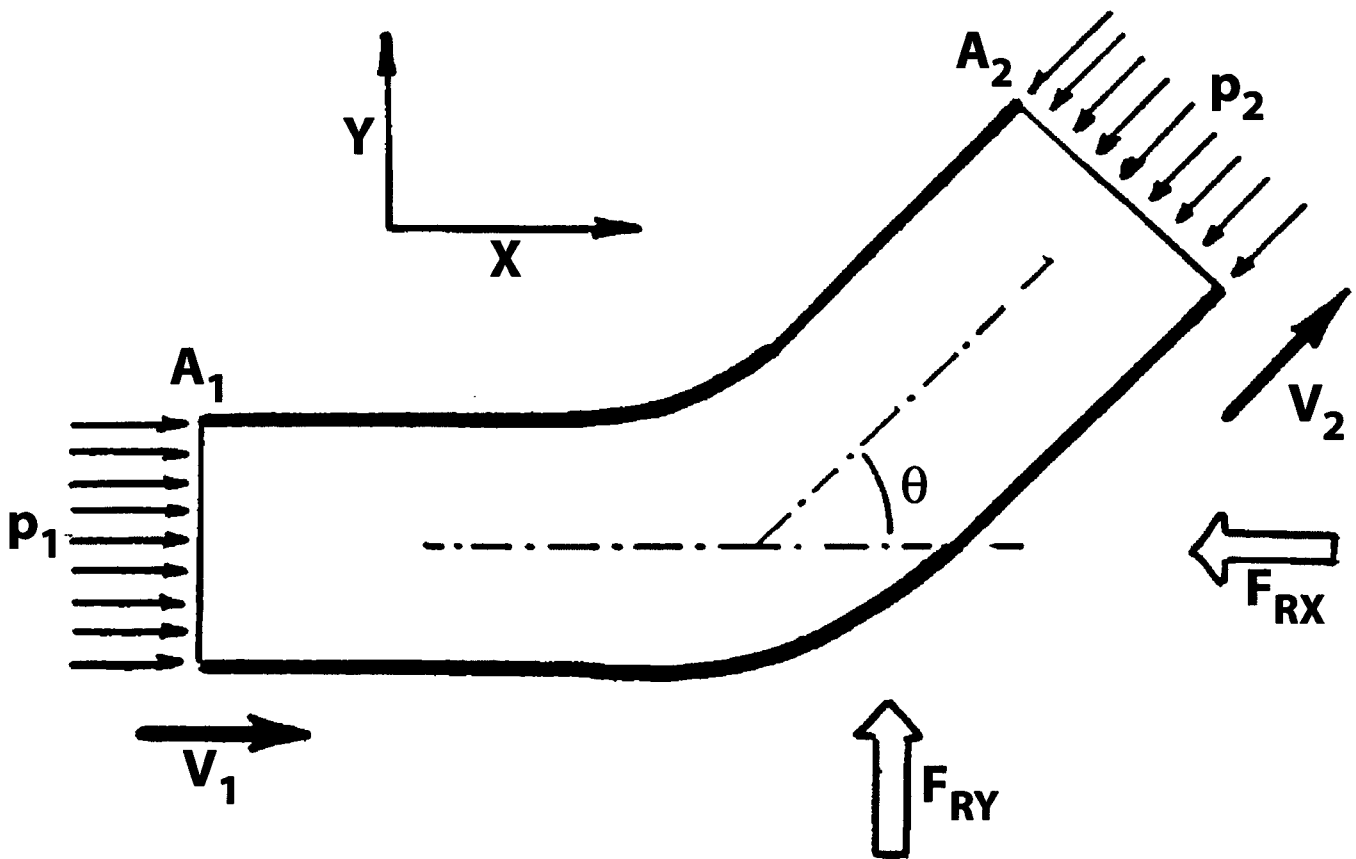
### Reaction

$$* \quad F_R = p_1 A_1 - p_2 A_2 - \rho Q (V_2 - V_1)$$

\* Note sign convention

# APPLICATIONS

## ANGLED PIPE



### Forces in X - Direction (on fluid)

$$\Sigma F_X = p_1 A_1 - p_2 A_2 \cos \theta - F_{RX} = \rho Q \Delta V_X$$

### Reaction

$$F_{RX} = p_1 A_1 - p_2 A_2 \cos \theta - \rho Q (V_2 \cos \theta - V_1)$$

### Forces in Y - Direction (on fluid)

$$\Sigma F_Y = - p_2 A_2 \sin \theta + F_{RY} = \rho Q \Delta V_Y$$

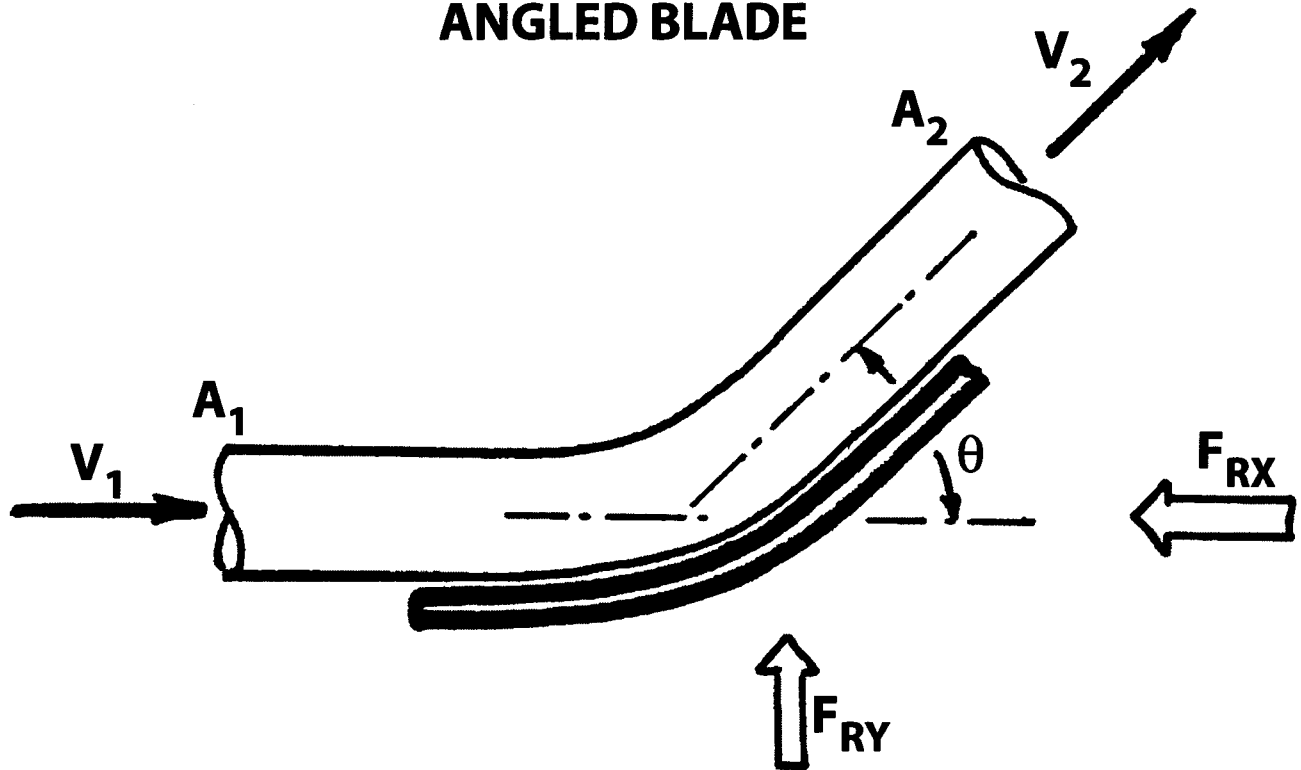
### Reaction

$$F_{RY} = p_2 A_2 \sin \theta + \rho Q V_2 \sin \theta$$

# FORCES IN FREE JETS

## MOMENTUM EQUATION APPLIED TO BLADES

### ANGLED BLADE



**Use Previous Equation for Flow in Pressure Conduit**

**Reaction in X - Direction**

$$F_{RX} = p_1 A_1 - p_2 A_2 \cos \theta - \rho Q (V_2 \cos \theta - V_1)$$

**Since  $p_1$  and  $p_2$  are zero**

$$F_{RX} = -\rho Q (V_2 \cos \theta - V_1)$$

**Reaction in Y - Direction**

$$F_{RY} = p_2 A_2 \sin \theta + \rho Q V_2 \sin \theta$$

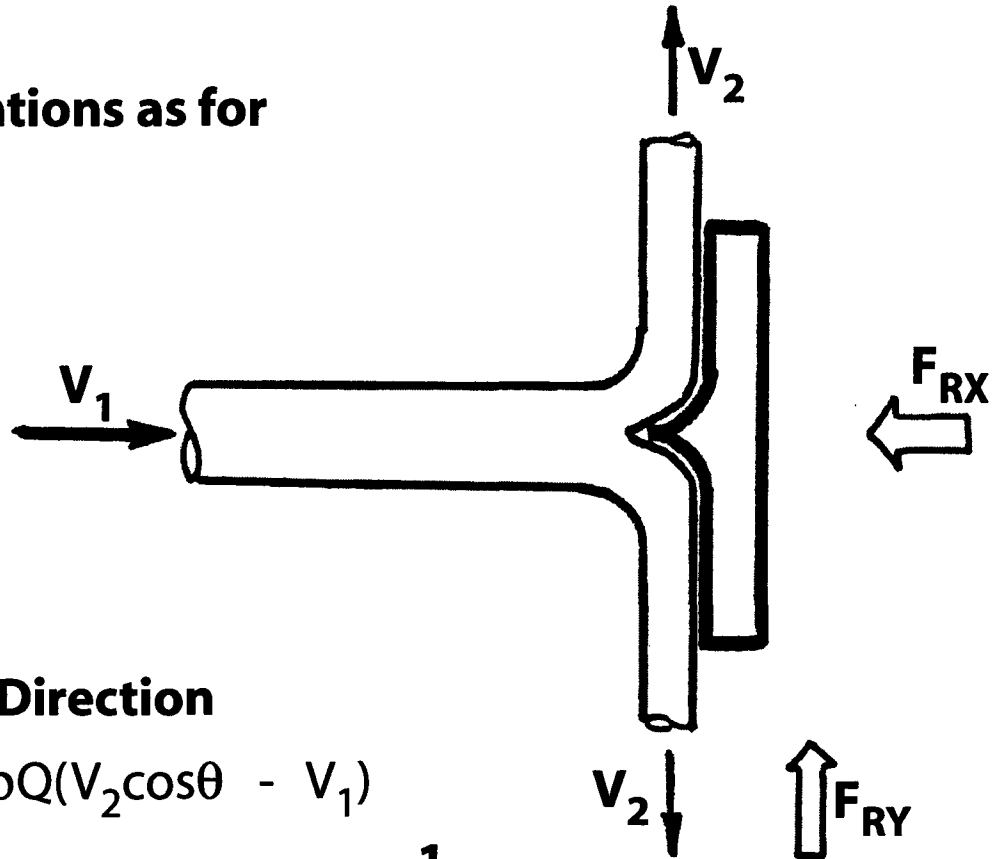
**Since  $p_1$  and  $p_2$  are zero**

$$F_{RY} = \rho Q V_2 \sin \theta$$

# FORCES IN FREE JETS

## FLAT BLADE

Use Same Equations as for Angled Blade



### Reaction in X - Direction

$$F_{RX} = -\rho Q(V_2 \cos\theta - V_1)$$

For Equally Divided Flow  $Q_2 = \frac{1}{2}Q_1$

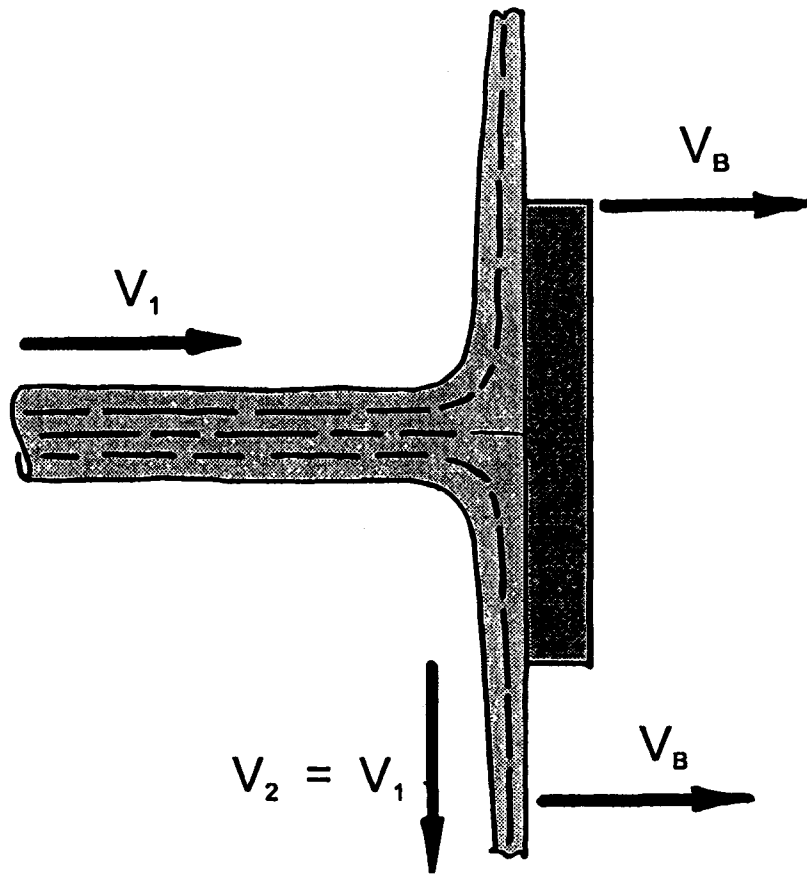
$$\begin{aligned} F_{RX} &= -\rho \frac{1}{2}Q(-V_1) - \rho \frac{1}{2}Q(-V_1) \\ &= \rho QV_1 \end{aligned}$$

### Reaction in Y - Direction

$$F_{RY} = \rho QV_2 \sin\theta$$

For Equally Divided Flow  $Q_2 = \frac{1}{2}Q_1$

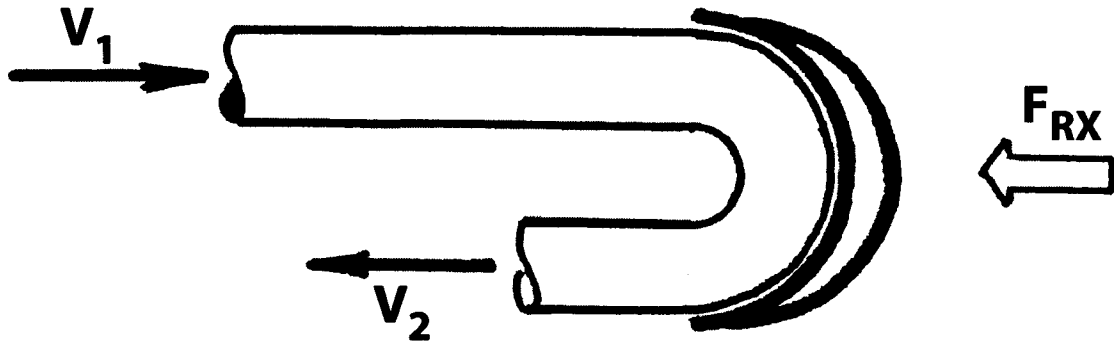
$$\begin{aligned} F_{RY} &= \rho \frac{1}{2}QV_2 - \rho \frac{1}{2}QV_2 \\ &= 0 \end{aligned}$$



**Figure 1 Jet impinging upon a flat moving plate**

# FORCES IN FREE JETS

## U-BLADE



**Use Previous Equation**

**Reaction in X - Direction**

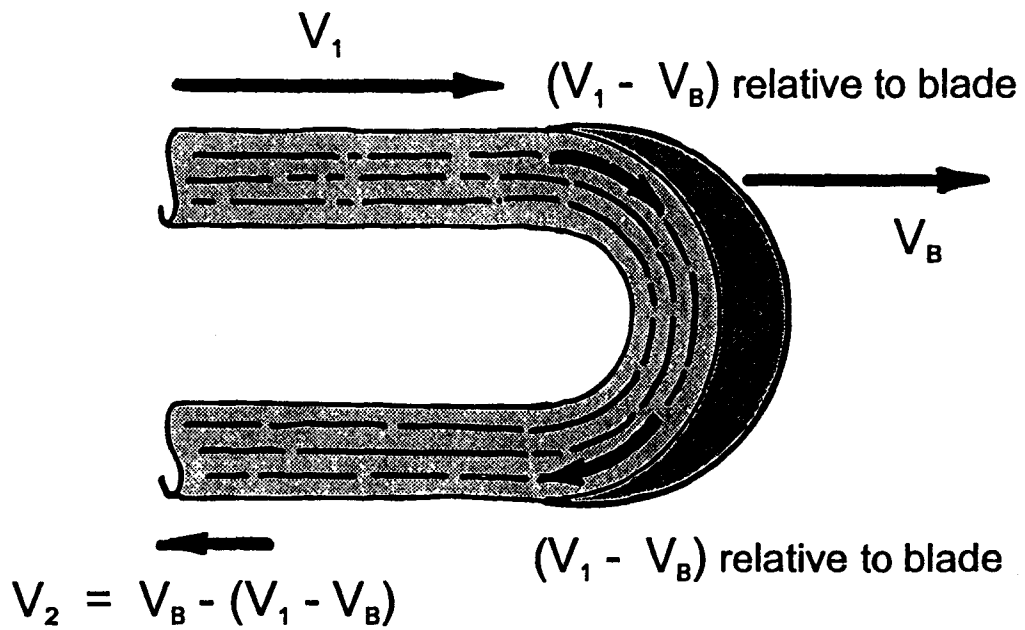
$$F_{RX} = -\rho Q(V_2 \cos\theta - V_1)$$

Note that  $\cos 180^\circ = -1$

$$\begin{aligned} F_{RX} &= -\rho Q(-V_2 - V_1) \\ &= \rho Q(V_2 + V_1) \end{aligned}$$

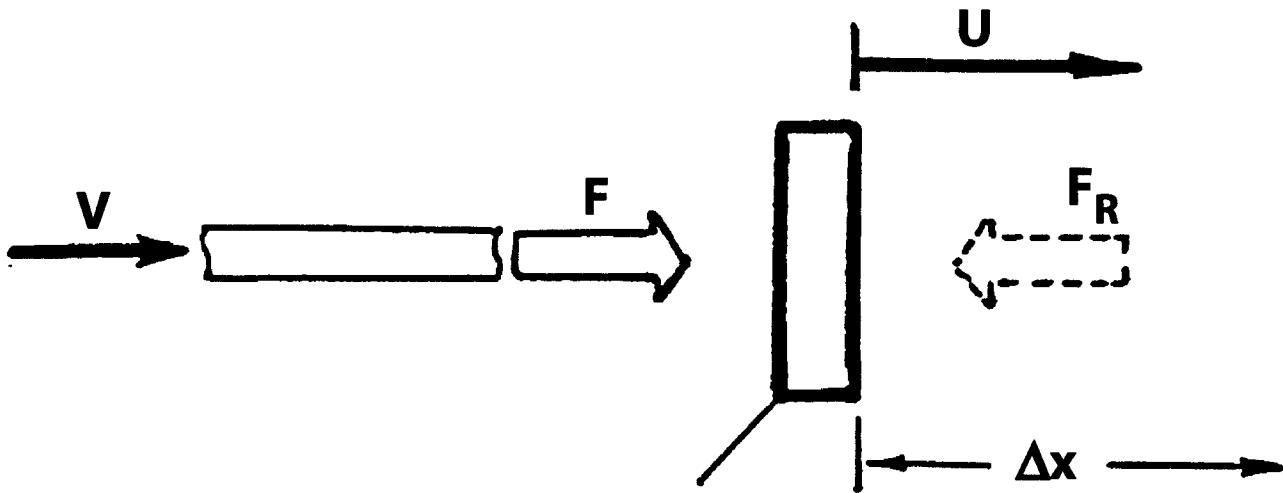
Note that the Energy Equation gives  $V_2 = V_1$  for no friction

$$F_{RX} = \rho Q 2V_1$$



**Figure 2 Jet impinging upon a curved moving plate**

# WORK AND POWER



**Force on blade**       $F = \rho Q \Delta V$       (N)

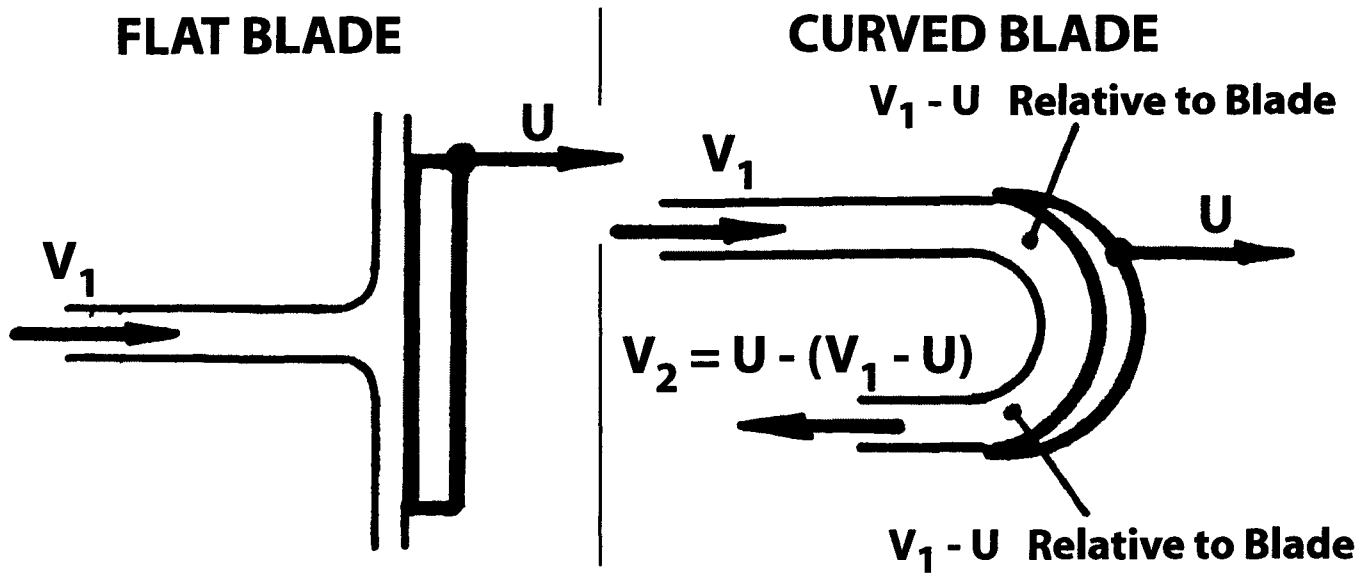
**Work on blade**       $W = \text{Force} \times \text{Distance}$   
 $= F \Delta x$   
 $= \rho Q \Delta V \Delta x$       (Nm)

**Power to blade**       $P = \text{Work} / \text{Time}$   
 $= F \Delta x / \Delta t$   
 $= F U$   
 $= \rho Q \Delta V U$       (J/s)

Now since       $\Delta V = (V - U)$

**Power to blade**       $P = \rho Q (V - U) U$

# IMPULSE POWER



## Force on Blade

$$F = \rho Q(\Delta V)$$

$$= \rho Q(V_1 - U)$$

$$F = \rho Q(\Delta V)$$

$$= \rho Q[V_1 - \{U - (V_1 - U)\}]$$

$$= \rho Q[V_1 - \{U - V_1 + U\}]$$

$$= \rho Q[2V_1 - 2U]$$

$$= \rho Q2(V_1 - U)$$

## Power to Blade

$$P = \rho Q(V_1 - U)U$$

$$= \rho Q(V_1 U - U^2)$$

$$P = \rho Q2(V_1 - U)U$$

$$= \rho Q2(V_1 U - U^2)$$

## Blade Speed for Maximum Power $dP/dU = 0$

$$\frac{dP}{dU} = \rho Q(V_1 - 2U)$$

$$0 = \rho Q(V_1 - 2U)$$

$$V_1 = 2U$$

$$\frac{dP}{dU} = 2\rho Q(V_1 - 2U)$$

$$0 = 2\rho Q(V_1 - 2U)$$

$$V_1 = 2U$$

### For Maximum Power

$$U = \frac{1}{2}V_1$$

$$U = \frac{1}{2}V_1$$

### Maximum Power on Blade

$$P = \rho Q(V_1 - \frac{1}{2}V_1) V_1$$

$$P = \rho Q 2(\frac{1}{2}V_1) \frac{1}{2}V_1$$

$$= \rho Q(\frac{1}{2}V_1) \frac{1}{2}V_1$$

$$= \rho Q 2(\frac{1}{2}V_1) \frac{1}{2}V_1$$

$$= \frac{1}{4}\rho Q V_1^2 \quad (\text{J/s})$$

$$= \frac{1}{2}\rho Q V_1^2 \quad (\text{J/s})$$

### Jet Power

$$E = \frac{1}{2}M V_1^2$$

$$E = \frac{1}{2}M V_1^2$$

$$= \frac{1}{2}\rho Q V_1^2 \quad (\text{J/s})$$

$$= \frac{1}{2}\rho Q V_1^2 \quad (\text{J/s})$$

### Efficiency = Output / Input

$$\eta = P / E$$

$$\eta = P / E$$

$$= \frac{1}{4}\rho Q V_1^2 / \frac{1}{2}\rho Q V_1^2$$

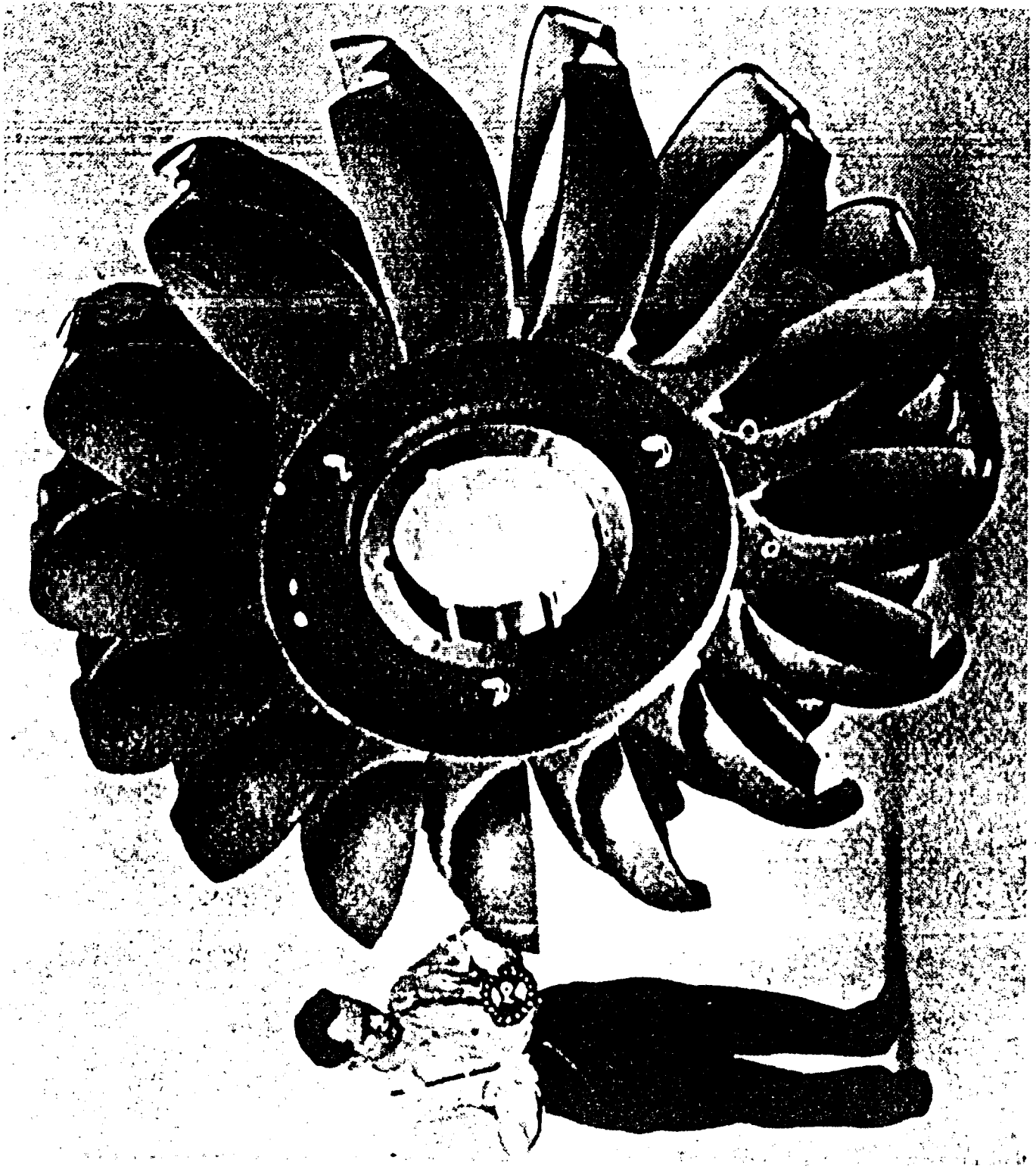
$$= \frac{1}{2}\rho Q V_1^2 / \frac{1}{2}\rho Q V_1^2$$

$$= \frac{1}{2}$$

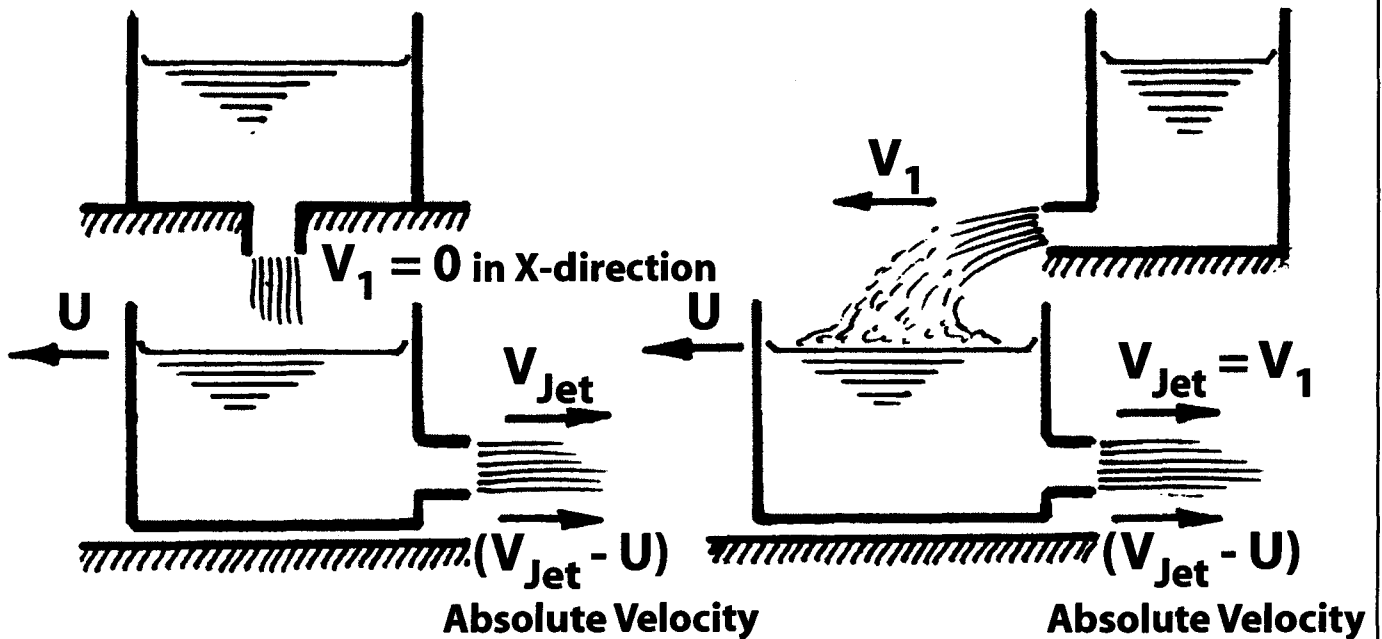
$$= 1$$

$$= 50\%$$

$$= 100\%$$



# REACTION POWER



## Force on Bucket

$$\begin{aligned}
 F &= \rho Q(\Delta V) \\
 &= \rho Q[0 - (V_{\text{Jet}} - U)] \\
 &= \rho Q(-V_{\text{Jet}} + U)
 \end{aligned}$$

$$\begin{aligned}
 F &= \rho Q(\Delta V) \\
 &= \rho Q[-V_1 - (V_{\text{Jet}} - U)] \\
 &= \rho Q(-V_1 - V_1 + U) \\
 &= \rho Q(-2V_1 + U)
 \end{aligned}$$

## Power to Bucket

$$\begin{aligned}
 P &= F(-U) \\
 &= \rho Q(-V_{\text{Jet}} + U)(-U) \\
 &= \rho Q(V_{\text{Jet}} - U)^2
 \end{aligned}$$

$$\begin{aligned}
 P &= F(-U) \\
 &= \rho Q(-2V_1 + U)(-U) \\
 &= \rho Q(2V_1U - U^2)
 \end{aligned}$$

## Bucket Speed for Maximum Power $dP/dU = 0$

$$\frac{dP}{dU} = \rho Q(V_{\text{Jet}} - 2U)$$

$$\therefore U = \frac{1}{2}V_{\text{Jet}}$$

$$\frac{dP}{dU} = \rho Q(2V_1 - 2U)$$

$$\therefore U = V_1$$

## For Maximum Power

$$U = \frac{1}{2}V_{\text{Jet}}$$

$$U = V_1 = V_{\text{Jet}}$$

## Maximum Power on Bucket

$$P = \rho Q(-V_{\text{Jet}} + \frac{1}{2}V_{\text{Jet}})(-\frac{1}{2}V_{\text{Jet}})$$

$$= \rho Q(-\frac{1}{2}V_{\text{Jet}})(-\frac{1}{2}V_{\text{Jet}})$$

$$= \frac{1}{4}\rho QV_{\text{Jet}}^2$$

$$P = \rho Q(2V_1 + V_1)(-V_1)$$

$$= \rho Q(-V_1)(-V_1)$$

$$= \rho QV_1^2$$

## Potential Jet Power in Direction of Motion

$$E = \frac{1}{2}\rho QV_{\text{Jet}}^2$$

$$E = \frac{1}{2}\rho QV_{\text{Jet}}^2 + \frac{1}{2}\rho QV_1^2$$

$$= \frac{1}{2}\rho QV_1^2 + \frac{1}{2}\rho QV_1^2$$

$$= \rho QV_1^2$$

## Efficiency = Output / Input

$$\eta = P / E$$

$$= \frac{1}{4}\rho QV_{\text{Jet}}^2 / \frac{1}{2}\rho QV_{\text{Jet}}^2$$

$$= \frac{1}{2}$$

$$= 50\%$$

$$\eta = P / E$$

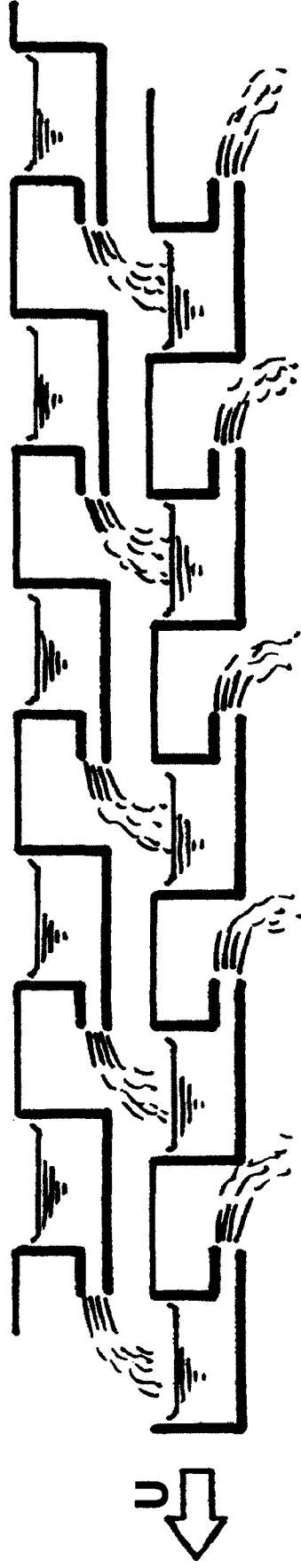
$$= \rho QV_1^2 / \rho QV_1^2$$

$$= 1$$

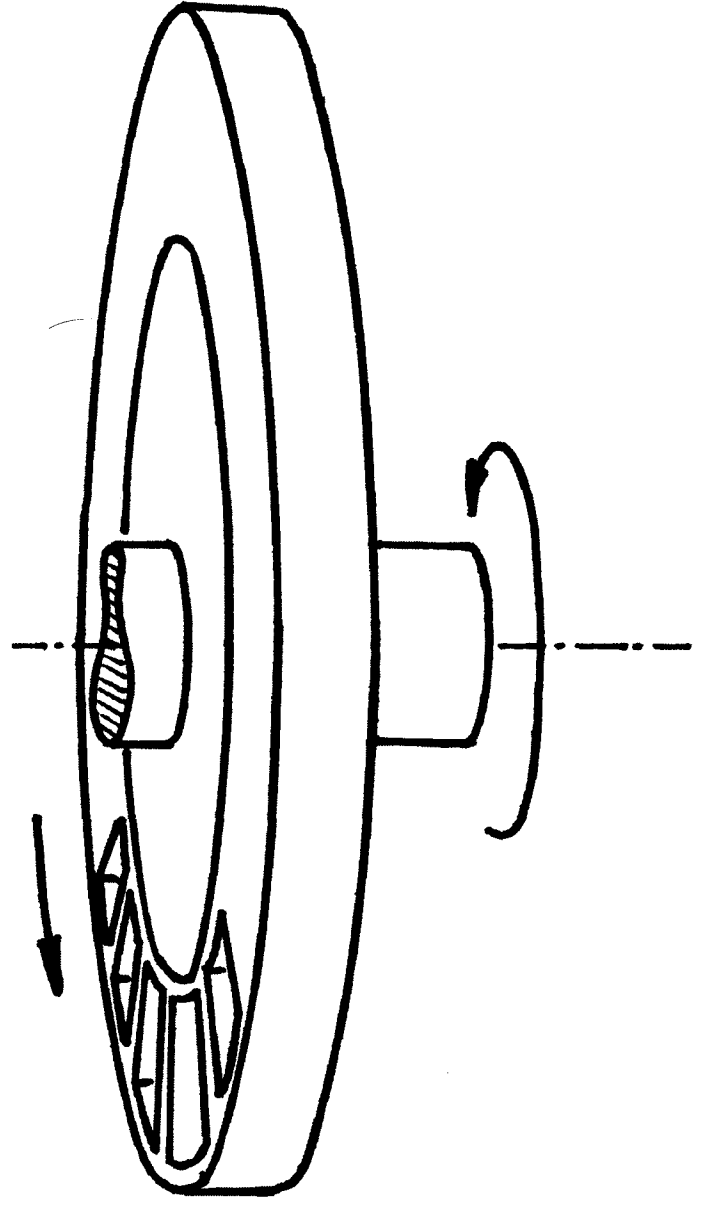
$$= 100\%$$

# REACTION TURBINE

Sequence of Fixed and Moving Buckets



Circular Arrangement of Buckets





**Figure 16.3** Francis runner at  
Niagara Falls.  $h = 214$  ft.  
 $n = 107$  rpm.  
 $\eta = 93.8$  percent at 72,500 hp,  
diameter = 176 in, overall  
diameter at band =  $183\frac{3}{8}$  in.  
(Courtesy of Allis-Chalmers  
Mfg. Co.)